TEST AND EVALUATION OF A PILOT TWO — STAGE PRECIPITATOR FOR JET ENGINE TEST CELL EXHAUST GAS CLEANING

AT

NAVAL AIR REWORK FACILITY NAVAL AIR STATION JACKSONVILLE, FLORIDA

DISTRIBUTION STATEMENT A

Approved for public release;
Distribution Unlimited

THE POLICE A constructors inc.

Best Available Copy



DEPARTMENT OF THE NAVY SOUTHERN DIVISION NAVAL FACILITIES ENGINEERING COMMAND 2144 MELBOURNE ST., P. O. BOX 10068 CHARLESTON, S. C. 29411

PLEASE ADDRESS REPLY TO THE CUMMANDING UPPICES, NOT TO THE SIGNES OF THIS LETTER-

Code 406 17 September 1976

Defense Documentation Center Attention TCA, LaChance Cameron Station, Bldg. 5 Alexandria, Virginia 22314

Gentlemen:

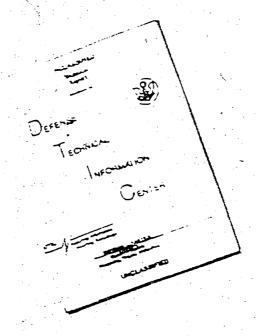
Attached is the release you requested in your telephone conversation of 20 July 1976 with the undersigned.

Yours_very truly,

Head Specifications Branch



DISCLAIMER NOTICE



THIS DOCUMENT IS BEST QUALITY AVAILABLE. THE COPY FURNISHED TO DTIC CONTAINED A SIGNIFICANT NUMBER OF PAGES WHICH DO NOT REPRODUCE LEGIBLY.

REPRODUCED FROM BEST AVAILABLE COPY



September 14, 1976

Commanding Officer
Southern Division
Naval Facilities Engineering Command
2144 Melbourne Street
Charleston, South Carolina 29411

Attention: Code 406

Gentlemen:

Test and Evaluation of a Pilot Two-Stage Precipitator for Jet Engine Test Cells Contract N62467-74-C-0161

Attached is one copy of a letter from American Air Filter Co. authorizing reproduction of AAF Drawing 835-A which is contained in our report on the Pilot Precipitator Evaluation.

Very truly yours,

Dall Uh

D. G. Munson

DGM:bd

RN: 6183-003

Enc.

Mr. James A. Ferner Project Manager United Engineers & Constructors, Inc. 600 Park Square Building Boston, Massachusetts 02116

Dear Mr. Ferner:

Permission is hereby granted to reproduce American Air Filter Company Drawing DEV-835A, in line with your request to our Mr. John Ashe.

Please b: sure that AAF receives credit for this drawing.

Sinceraly.

Robert C. Braverman

RCB/jh

Encl.

cc -J. T. Ashe C. J. Bressoud

- ACCESSION	tor	
ATIS .	While Section	
800	Buff Section	
UMANAGUM	(ED	U
PY	HX. ON GIL	
Pist.	AVAIL and/or SPEE	TAL
A		

TEST AND EVALUATION OF A
PILOT TWO – STAGE PRECIPITATOR
FOR JET ENGINE TEST CELL
EXHAUST GAS CLEANING

Prepared for:

NAVAL AIR REWORK FACILITY

NAVAL AIR STATION

JACKSONVILLE, FLORIDA

APR**Æ ₽**76

YII

10) 1 N621/11-12-2-07(1) 1 NOUND 5- 70-C-0021

R. N. 6188 - 003

DDC

DECILITATE

JUN 22 1976

NEWESTUE

D

PRINCIPAL CONTRIBUTORS TO THIS REPORT:

J. A. FERNER – PROJECT MANAGER
D. G. MUNSON – MECHANICAL ENGINEER

DISTRIBUTION STATEMENT A

Approved for public release; Distribution Unlimited 357800



April 26, 1976

Department of the Navy Southern Division Naval Facilities Engineering Command Charleston, South Carolina 29411

Gentlemen:

Test and Evaluation of an
Electrostatic Precipitator for
Jet Engine Test Cells
Naval Air Rework Facility
Jacksonville, Florida
Phase II - Contract N62467-74-C-0161

We are pleased to submit herewith our report on the Test and Evaluation of a Two-Stage Precipitator used for Jet Engine Test Cell Exhaust Gas Cleaning. The efficiency tests were conducted on a prototype of the precipitator installed at the Black Point Test Cell, Naval Air Rework Facility, Jacksonville, Florida.

We conclude that the precipitator will operate satisfactorily in the environment of the test cell exhaust stack and that its particulate removal capability is comparable to that of the crossflow scrubber concept now being applied to test cell exhaust gas cleaning. Capital costs for a precipitator system providing this performance are estimated at \$1.40-\$1.70 per ACFM of test cell exhaust flow depending on test cell size. Direct operating costs are estimated at \$65-\$130 per engine test depending on engine size.

Should the overall owning and operating costs of the precipitator concept compare favorably with those of the crossflow scrubber, we recommend that final performance testing of the prototype be completed using the EPA Method 5 technique. This should be done prior to a decision on full scale application.

We wish to acknowledge the assistance and participation of personnel associated with the following organizations:

1

Southern Division - Naval Facilities Englueering Command,
Charleston, South Carolina
Naval Air Rework Facility - Jacksonville, Florida
American Air Filter Co., Inc. - Louisville, Kentucky

This report has been prepared under Naval Facilities Engineering Command Contract N62467-74-C-0161.

Very truly yours,

J. A. Ferner Project Manager

Approved by:

J. H. Fullerton Vice-President

General Engineering Division

JAF:bd

RN: 6183-003

TABLE OF CONTENTS

Introduction

Section 1	Scope of Work
Section 2	Summary and Conclusions
Section 3	Discussion of Test Program and Procedures
Section 4	Results of Performance Testing
Section 5	Full-Scale System Description and Economics
Section 6	APPENDIX
Section 7	BIBLIOGRAPHY

INTRODUCTION

NAVFACENGEOM Contract NOO0025-72-C-0037 to study available means for the abatement of air pollution caused by operation of Naval jet engine test facilities. The findings of the study, issued in August 1973, were that the use of fuel additives, the retrofit of smokeless combustors and the installution of gas cleaning equipment were potential means of controlling particulate emissions from the cells. Additives and smokeless combustors were found to require additional development leaving exhaust gas cleaning as the only technology then available for emission control. A two-stage electrostatic precipitator was recommended as the most viable alternative to a concept then being actively developed, the cross-flow wet scrubber.

Due to the unique nature of the application and the high cost of full-sized equipment, it was recommended that a bench scale precipitator be tested to confirm performance and establish size parameters. Such a prototype unit was subsequently installed at Black Point test cell No. 1, Naval Air Rework Facility, Jacksonville, Florida and underwent a sequence of performance and operating tests under the supervision of UE&C.

This report summarizes the results of the test program and provides data on the economics of applying a full-scale system 'o a jet engine test cell.

√/

SECTION 1

SCOPE OF WORK

1.01 This report summarizes the results of the test program, offers an evaluation of those results and discusses their implication with regard to the technical and economic feasibility of applying the two-stage electrostatic precipitator to test cell exhaust gas cleaning.

Specific objectives of the test program were as follows:

- Confirm that the equipment will operate satisfactorily in the environment of the test cell exhaust stack.
- Establish the maximum gas velocity and minimum field depth at which the equipment will meet performance requirements.
- Document equipment performance using source testing techniques established as acceptable by the Environmental Protection Agency for the particular source application.

 These techniques are also identical to those used in testing the cross-flow scrubber concept and thereby should produce results which are directly comparable.
- Establish operating cost parameters for the equipment.
- Establish capital cost parameters for equipment.

SECTION 2

SUMMARY AND CONCLUSIONS

2.01 Summary

- 2.01.1 Gas sampling tests were conducted on a prototype twostage precipitator installed to clean a portion of the exhaust gases from Black Point Cell No. 1 at the Naval Air Rework Facility, NAS, Jacksonville, Florida. The purpose of the test program was to determine the feasibility of full scale application of the precipitator concept to jet engine test cell exhaust gas cleaning.
 - 2.01.2 Equipment testing proceeded in three phases:
- Tests at various exhaust gas throughputs using the equipment manufacturer's standard test method. These tests were conducted for the purpose of establishing design parameters for the equipment.
- Tests at the established design gas throughput using the standard EPA Method 5 test procedure. These tests were conducted for the purpose of documenting equipment performance using test techniques acceptable to regulatory authorities. A second test team, using different test equipment and procedures, performed simultaneous tests in order to obtain correlation between test techniques.
- A repeat of the manufacturer's tests following a period of sustained operation of the equipment. The purpose of repeat testing was to detect any degradation in equipment performance with use.
- 2.01.3 Results of the initial series of tests performed at the established design conditions are summarized in Table 2-1.

TABLE 2.1

Test Dates	Test Methods	Inlet Concentration (GR/DSCF)	Outlet Concentration (GR/DSCF)	Efficiency (% Removed)
4-14-75 & 4-15-75	Mfg's. Test Method	41-43 X 10 ⁻⁴	3-4 x 1.0 ⁻⁴	91-93
4-17-75 & 4-18-75	EPA Method 5	32-73 X 10 ⁻⁴	12-35 X 10 ⁻⁴	51-65
4-17-75 & 4-18-75	Modified EPA inlet; beta analyzer outlet *	8-9 X 10 ⁻⁴	4-7 X 10 ⁻⁴	17-55

^{*}Conducted concurrently with EPA Method 5 tests.

2.01.4 Testing of the equipment utilizing the manufacturer's standard method indicated no degradation in equipment performance after approximately 5 weeks of normal operation. Degradation tests using the EPA Method 5 have not been conducted to date due to nonavailability of the test cell.

2.02 Conclusions

- 2.02.1 The two-stage precipitator will operate satisfactorily in the hot wet environment of a jet engine test cell exhaust stack. A mist eliminator should be installed upstream of the equipment to prevent liquid carry over from the evaporative cooling system.
- 2.02.2 The lack of correlation in the results of equipment efficiency testing using the manufacturer's procedure and the EPA Method 5 seem attributable to inherent differences in the test methods since data obtained on multiple runs using the same technique are in fair agreement.

In determining compliance with air pollution regulations, the EPA Method 5 data should be used since this data was obtained using equipment and procedures specified by most regulatory authorities.

Precipitator and the cross-flow scrubber are similar. The precipitator removed an average of 59% of the incoming particulate according to the EPA Method 5 tests. The cross-flow scrubber exhibited a 55% average removal efficiency during an extensive series of tests conducted in conjunction with an earlier test program. The latter testing utilized the same engine, test configuration and methodology as the precipitator testing. Both efficiency figures refer to the particulate removed in the control equipment proper and do not include removal by the spray system. On the basis of the above comparisons, the performance of the precipitator appears to be at least equal to that of the cross-flow scrubber.

2.02.4 The combined efficiency of the evaporative cooling system, which acts as a prescrubber removing 50-60% of the particulate emission, and the two-stage precipitator averaged 86% on the three EPA Method 5 tests. Emissions leaving the equipment were well below any established standard.

2.02.5 Capital costs for a pollution abatement system incorporating the precipitator concept would be on the order of \$850,000 for a 500,000 AGFM system or \$1,690,000 for a 1,200,000 AGFM system based on a 500 FPM design velocity.

2.02.6 Annual operating costs for a pollution abatement system incorporating the precipitator concept can be on the order of \$32,000 for a 500,000 ACFM system testing 500 J-79 engines per year or \$65,000 for a 1,200,000 ACFM system testing 500, 350 lb/sec engines per year.

2.03 Recommendation

2.03.1 The prototype precipitator and test cell duct is still in place at the Black Point Test Cell. However, the test cell stack has been extensively rebuilt without provision for the test duct penetration and thus additional work would be required prior to resumption of the testing.

The results of the initial testing with EPA Method 5 procedures indicate that the two-stage precipitator and the cross-flow wet scrubber are comparable from the standpoint of performance. We feel that the next logical step should be a comparison of the capital and operating costs of the two systems. Costs associated with the two-stage precipitator concept, estimated on the basis of parameters developed during the test program, are given in this report for two test cell sizes. Cost factors for the cross-flow scrubber should be available from the systems now being installed at the Jacksonville and Norfolk Naval Air Rework Facilities.

Should life cycle costs of the precipitator concept compare favorably with those of the wet scrubber, final performance testing of the prototype should be completed to document performance to the satisfaction of regulatory authorities.

2.03.2 It is evident that test cell exhaust gas cleaning will be an expensive proposition regardless of the type of control equipment installed. Continued work on alternative measures, fuel additives in the near term and clean burning engines in the far term, is certainly warranted. Application of exhaust gas cleaning should be limited to specific cells scheduled to test older engines which, for reasons of performance, cannot use fuel additives.

SECTION 3

DISCUSSION OF TEST PROGRAM

AND PROCEDURES

The test program was divided into five phases which are summarized below and outlined in detail in Reference 4.

Phase I - Manpower and Equipment Scheduling - Two test teams took part in the program. One team, staffed by American Air Filter Inc. the supplier of the prototype, calibrated the exhaust gas draw-off apparatus and performed the type "A" tests described below. A second team, staffed by Jacksonville Naval Air Rework Facility personnel, performed the type "B" tests also described below.

A third test team, staffed by personnel from the Aircraft Environmental Support Office of the Naval Environmental Protection Support Service (NEPSS), conducted tests simultaneously with the type "B" tests. These tests were not a part of the NAVFAC test program but were run for the purpose of comparing results obtained by the NARF Jax team with those obtained by the NEPSS team which used a different type of sampling equipment.

Phase II - Equipment Checkout and Pretest Calibration - The pretest calibration phase consisted of calibrating the venturi section differential to the exhaust gas flow through the test system as measured by a pitot tube traverse. The venturi reading would then be used to compute gas flow rate during the type "A" testing to determine optimum velocity.

The procedure, as originally constituted, called for the utilization of a throttling damper installed at the outlet of the prototype to vary gas flow. However, the unexpectedly high kinetic energy of the

exhaust gas entering the test duct rendered this procedure unworkable due to the inability of the ductwork to contain the static pressure developed by the throttling action.

As a result of the above, it was necessary to vary the gas flow rate through the test duct by means of varying the orifice area at the inlet of the duct section. New pitot tube traverse holes were drilled in the 1 ft. x l ft. sections of the duct upstream of the venturi section in order to produce higher and therefore more accurate velocity head readings. Four calibration runs were performed using orifice areas of 12 in.², 18 in.², 24 in.² and 36 in.². The orifice area vs gas flow relationship provied to be linear producing test flows of 2517, 3826, 4732 and 7334 ACFM respectively.

Phase III - Initial Performance Testing - This phase of the test program, conducted during the period of April 7-18, 1975 and June 4-6, 1975, included both type "A" and type "B" efficiency tests. The type "A" testing served the dual purposes of establishing equipment compatibility with the test cell exhaust gas and determining the maximum exhaust gas throughput velocity at which the unit would operate with a satisfactory collection efficiency. The type "B" tests were conducted to document equipment performance and, additionally, to provide data which could be directly compared to data accumulated during a previous test program associated with the cross-flow wet scrubber prototype.

3.03.1 Type "A" Test - A total of eight (8) type "A" tests were run during the initial performance phase. The type "A" tests utilized American Air Filter's standard test procedure which called for single-point sampling at the inlet and outlet of the prototype. The average velocity through the ductwork (as determined by the pretest calibration runs) was

used to establish sampling rates which would approximate isokinetic conditions at the sample probe nozzle.

Tests were conducted at each of the four operating points corresponding to gas velocities of 172, 262, 323, and 501 FPM through the prototype. Unit efficiency was calculated after each run utilizing NARF laboratory facilities. Using 90% by weight collection efficiency as performance criteria, the maximum allowable velocity through the prototype was established as approximately 500 FPM.

3.03.2 Type "B" Tests - A total of seven (7) type "B" tests were run during the initial performance phase. Three tests were performed immediately following the type "A" tests and utilized the same J-79 engine. The remaining tests were performed approximately six (6) weeks after the initial tests.

The type "B" tests allowed computation of both particulate removal by the prototype and particulate removal by entrainment in the unevaporated portion of exhaust gas cooling water. The standard EPA (3) Method 5 procedure was utilized to determine particulate removal from the gas stream. A 20-point grid was sampled at the inlet and outlet of the unit with sample flow rate adjusted at each point to produce isokinetic conditions at the probe nozzle. Both solid and liquid particulates were collected and recorded. Particulate removal by water entruinment was determined by establishing the drain flow rate and concentration of particulates in the effluent. This concentration was then multiplied by the ratio of drain flow rate to exhaust gas flow rate to obtain a number comparable to the gas stream samples.

The procedures used in the type "B" testing conducted by NARF personnel duplicated those used by the same personnel in testing the wet crossflow scrubber model (2).

3.03.3 NPESS Tests - The NEPSS tests were conducted concurrently with the type "B" tests run April 17-18, 1975. Inlet sampling was conducted using an Aerotherm high volume EPA Method 5 particulate sampler which is similar to a standard EPA train that has been scaled up in size to allow high volume sampling. Outlet sampling was conducted using a Lear Seigler PM/Argos I continuous particulate mass emission analyzer. This device measures the attenuation of beta radiation by particulate collected on a filter tape and converts this to a measure of the particulate mass.

3.04 <u>Phase IV - Normal Operating Runs</u> - The purpose of this phase of the program was to obtain an indication of equipment durability under normal operating conditions. The prototype was operated during normal engine testing sequences between the dates of June 6, 1975 and July 13, 1975.

Phase V - Final Performance Testing - Performance testing was conducted following the normal operating period in order to detect any degradation in equipment performance with time. Two type "A" tests were run on July 15, 1975 at the design velocity of 500 FPM through the precipitator. The performance indicated by these tests was essentially the same as that indicated by the initial service of type "A" tests.

This phase of the test program also called for additional type "B" testing in order to obtain "degradation" data using testing techniques acceptable to pollution control authorities. However, to date, the Black Point cell has not been available to support such tests due to the heavy schedule of production engine tests and extensive modification of the test cell stack.

SECTION 4

RESULTS OF EQUIPMENT TESTING

- 4.01 Equipment Reliability In general, the equipment proved capable of operating satisfactorily when handling the test cell exhaust gas.

 Several operating problems did, however, occur during the course of the test program indicating design modifications recommended below.
- 4.01.1 Excessive Arcing The location of the exhaust gas draw-off duct was such that a large amount of cooling water was entrained in the gas entering the prototype. This caused a substantial amount of arcing within the precipitator. Excessive arcing is detrimental to equipment operation in three ways:
 - (1) Short circuiting occurs between collecting plates thus reducing the effective strength of the electrostatic field between plates. Since it is this field that forces the ionized particulates towards the collecting plates, the overall effect of reducing the strength of this field is reduced particle deposition and reduced collection efficiency.
 - (2) Average electric power consumption is increased and high peak power inputs induced by the short circuiting are encountered.
 - (3) The power peaks occassionally cause the circuit protection equipment contained in the precipitator control units to shut the equipment down.

Although entrained moisture should be less of a problem at the top of the test cell stack, it is recommended that any full scale installation be equipped with a mist eliminator upstream of the precipitator. 4.01.2 <u>Failure of High Voltage Leads</u> - Early in the test program during the velocity calibration runs, the equipment was tripped off the line several times due to failure of the insulation on the high voltage wiring supplying the ionizing and collecting electrodes. All failures occurred inside the precipitator casing where the wires were exposed to appreciable moisture.

The equipment was rewired using wire with a better grade of insulation (silicon-insulated) and no further problem was experienced. It is recommended that the high grade wire be specified for any full-scale installation.

- 4.01.3 Power Supply and Voltage Control Components A diode in one of the power pack assemblies failed during the initial testing phase and was replaced. Failure was apparently due to a manufacturing defect. No other problems were experienced with power pack components throughout the duration of the test program.
- 4.01.4 <u>Ionizer Wire Breakage</u> Two high voltage (12 kV) ionizers failed during the second series of type "A" tests. Replacement of ionizer wires is an item of routine maintenance discussed in Section 5.
- 4.02 Equipment Collection Efficiency Results of the type "A" and "B" tests are summarized in Table 4.1. Raw test data and details of test and data reduction procedures are given in Appendix A-1 for the type "A" tests and in Appendix A-2 and Reference 2 for the type "B" tests. NEPSS test results are reported separately in Reference 5.
- 4.02.1 Type "A" and "B" Tests Type "A" tests run between April 9 and April 14 established a maximum operating velocity through the unit at 500 FPM for a collection efficiency of approximately 90% and two collecting

TABLE 4.1

RESULTS OF EFFICIENCY TEST RUNS

	iyibe	Gas Flow	Velocity	Inlet	Outlet	Collecti	
		in Test Diet	thru	Part.	Part.	Prototype	Prototype & Spray Water
11		(ACFM)	(FPM)	(GR/DSCF)	(GR/DSCF)	(%)	(%)
	A	2517	172	0.0045	0.0008	82.22	
	Ą	2517	172	0.0018	0.0002	88.89	
	Ą	3828	262	0.0019	0.0001	94.74	
	Ą	4732	323	0.0031	0.0001	24.77	
	A	4732	323	0.003	0.0002	93.33	
	Ą	7334	501	0.0041	0.0003	92.68	
	A	7334	501	0.0043	0.0004	90.70	
	Ą	7334	501	0.0023	0.0005	78.3(1)	
	В	7148(2)	488.5(2)	0.00730(3)	0.00354(3)	51.5	
				0,00866(4)			**************************************
J-79 °	М	6622(2)	452.6(2)		0.00120 ⁽³⁾	6.49	
							90.73
J-79	ф	6980(2)	477(2)	0.00321(3)	0.00127 ⁽³⁾	4. 09	
				0.00822(4)			68.88
J-52	Д	6462(2)	441.6(2)		0.00344(3)	-80.1	
							2,82
			-				
	Best	Best Available Copy	le Copy				

Collection Efficiency type Prototype & Spray Water	(24)		33.52(1)		32.40 ⁽⁵⁾		31.6 (5)			
Collec Prototype	(2)	-23.08(1)		-25.00(5)		-19.6(5)		35.54	92.88	
Outlet Part. Conc	(GR/DSCF)	0.00192(3)		0.00240(3)		0.00238 ⁽³⁾		0.000272	0.000248	
Inlet Part. Conc	(GR/DSCF)	0.00156(3)	0.00155(4)	0.00192 ⁽³⁾	0.00163 ⁽⁴⁾	0.00199(3)	0.00149 ⁽⁴⁾	0.001882	0.003492	
Velocity thru Prototyne	(FPM)	445.8(2)		778.8(5)		451(2)		501	501	
Gas Flow in Test Duct	(ACFM)	6523(2)		6566(2)		6599(2)		7334	7334	
Test est est		m		ш		ĽΩ		 ¥	Ą	
rugine e		J-52		J-52		J-52		1-79	6½-f	
nate		-5-75		-6-75	- 	-6-75		-15-75	-15-75	

1) Tests run with one of two banks out of service.

Based on average of measured inlet and cutlet velocities. Measured inlet velocity exceeded measured outlet velocity on all "B" type tests except second run on 4-18-75. (2)

Includes solid particulates from probe and filter and liquid particulates from impingers. $\widehat{\mathcal{Z}}$

Particulates collected in sump drains normalized to a dry gas flow basis. Corrected for 10 ft2 duct.

Tests run with entire precipitator de-energized and out of service.

banks in series. Highest efficiency points were experienced at velocities in the 250-350 FPM range which was the anticipated design velocity range of the equipment. However, it was found that velocity could be increased to the 500 FPM range before efficiency fell off to 90%. The test of April 15 was run at 500 FPM with only one of the two series fields in service and indicated an efficiency of 78%. Ratioing this performance to the two-field performance, it was deduced that a single field in series would not meet performance criteria at any of the velocities tested. On the basis of the series of tests, equipment rat was determined by the manufacturer to be as follows:

Face velocity: 500 FPM

Field depth: Two cells

Efficiency: 90%

Type "B" testing conducted on April 17 and 18 indicated markedly lower efficiencies than the type "A" testing. Calculated precipitator efficiencies ranged from 51.1% to 64.9% averaging 59%. Total system efficiencies, which reflected particulates removed by spray water, ranged from 77.8% to 90.7% averaging 86%.

The data from the type "A" tests can be compared to the type "B" test data for the air sampling (Note 3 in Table 4.1), since both reflect particulate removal in the precipitator only. Inlet concentrations on the type "A" tests ranged from $18-45 \times 10^{-4}$ GR/DSCF which seem to be in reasonable agreement with the type "B" tests which had readings of 73, 39 and 32 x 10^{-4} GR/DSCF respectively. The principal differences in the data appear in the outlet concentrations. The type "A" data indicates 3-4 x 10^{-4} GR/DSCF (4-14-75 runs only) while the type "B" data shows markedly higher concentrations

of 12, 13 and 35 \times 10⁻⁴ GR/DSCF. Since testing was performed under essentially identical conditions, the differences in data apparently stem from the different testing techniques and equipment.

Since the type "B" tests utilize techniques accepted by the EPA and most local authorities and thereby will form the basis of documented equipment performance, NARF personnel undertook to conduct additional tests for the purpose of confirming the initial results. These tests were not run until June 4-6, 1975 due to engine testing requirements and mechanical problems with the Black Point cell.

The June 4 test indicated a precipitator efficiency of -80%. A subsequent test performed with only one field energized indicated an efficiency of -23% and two tests conducted with all power off indicated efficiencies of -25% and -19.6%. All tests were run with a J-52 engine as a pollution source in lieu of a J-79 engine.

Subsequent to the June testing, it was learned that during the interim period between the April and June tests, the blank-off plate which was installed at the sampling duct inlet had become dislodged. This allowed particulate-laden engine exhaust gas to pass through the prototype during normal engine production testing. Prolonged exposure to these exhaust gases could result in particulate buildup on the collecting plates to a point where they would begin to be re-entrained in the gas passing through the unit. This appears to be the only plausible explanation for the negative efficiency readings which are indicative of more particulate leaving the prototype than entering. Since normal operation of the prototype would encompass a washing cycle which would prevent particulate buildup on the plate, the June 4-6 data cannot be considered representative of equipment performance.

The final type "A" tests on July 15, 1975 were run after the prototype had been exposed to exhaust gases resultant from normal engine testing for a period of about 5 weeks. Two tests were run at design flow rate and indicated the same approximate level of performance as the initial test runs (89% efficiency).

As was mentioned earlier, a final series of type "B" tests were to be performed in this phase of the program. However, due to unavailability of the test cell, these tests have yet to be undertaken.

4.02.2 NEPSS Tests - The NEPSS tests were run concurrently with the type "B" tests of April 17-18, 1975. Inlet concentrations measured with the high volume EPA Method 5 train all measured in the range of 8-9 x 10^{-4} GR/SCF. These concentrations are substantially below those recorded in either the type "A" or "B" tests.

Other concentrations measured with the beta attenuation mass analyzer ranged from 4 to 7×10^{-4} GR/SCF. These values are below those recorded in the "B" tests and slightly above those recorded in the "A" tests. Overall precipitator efficiency calculated using the inlet and outlet concentrations averaged 38% ranging from 17% to 55%.

4.03 Compliance with Emission Regulations - No specific emissions standards have been established for jet engine test cells; however, a number of standards exist which are broad enough in scope to be considered applicable ⁽¹⁾. The most stringent of these regulations is the San Diego Air Pollution Control District Regulation limiting total particulate emissions (solid and liquid) to 0.1 GR/SCF with gas volume artifically corrected to a 12% CO₂ level. For JP-5 fuel, this is equivalent to 0.175 lbs/10⁶ BTU heat input. Using this emission rate as criteria and with available test data indicating particulate emissions in the range of 1.0

to 1.3 lbs/10⁶ BTU, the requirement for a 90% efficient abatement system was established.

Particulate emissions measured on the inlet side of the precipitator were an order of magnitude lower than those reported in previous test data. Inlet particulate loadings ranged from 0.04 lbs/10⁶ BTU to 0.09 lbs/10⁶ BTU in the type "A" tests which measured only solid particulates in the gas stream and from 0.24 lbs/10⁶ BTU to 0.33 lbs/10⁶ BTU in the type "B" tests on the J-79 which measured total particulates in both the gas and water streams. The emissions on the outlet side of the precipitator ranged from 0.002. lbs/10⁶ BTU to 0.017 lbs/10⁶ BTU in the type "A" tests and 0.025 lbs/10⁶ BTU to 0.073 lbs/10⁶ BTU on the type "B" tests. Thus, on all tests, emissions were below established standards.

It is not known where emissions from the test engine fall relative to emissions from the entire family of turbojet and turbofan engines. If the average efficiency for the initial type "B" tests of 36% is used as being indicative of system performance, the combined spray and precipitator systems would allow testing of engines emitting up to 1.25 lbs/10⁶ BTU without exceeding the San Diego Regulation.

4.04 Comparative Performance - Precipitator vs. Crossflow Scrubber - Both the proposed precipitator and crossflow scrubber system designs use a water quench to cool the test cell exhaust gases prior to treatment. This spray system acts as a prescrubber by entraining and removing particulate from the gases before they reach the control equipment. The type "B" tests, which measured particulate contained in the duct drains upstream of the precipitator and related them to the gas sampling data, indicated that from 55-70% of the total particulate was removed in this fashion. Identical tests

run on a model crossflow scrubber and the same J-79 engine during the period of November, 1973 to February, 1974 indicated approximately 60% removal in the same quench system.

A comparison of the gas sampling data which is representative of the particulate removed in the control equipment alone, indicates an average collection efficiency for the precipitator in the three type "B" tests of 59% (range 51.5% - 64.9%) and an average efficiency for the cross-flow scrubber of 55% (range 45% - 65%)⁽²⁾. Overall efficiency of the two systems computed on the basis of spray water and control unit removal averaged 86% for the precipitator and 78% for the crossflow scrubber.

On the basis of the above data, the performance of the two systems appears comparable.

4.05 Power Consumption - Power supply to the prototype was monitored at periodic intervals during testing. Excessive peaks were encountered with input current ranging from 4 to 8 amps at 120V AC and averaging approximately 6 amps. Average power input was, therefore, approximately 432 watts (.6 PF) or 0.059 KW/1000 ACFM. The power input was no doubt increased by the large amount of entrained moisture entering the unit. This effect should be rectified by the installation of the moisture eliminator discussed earlier.

4.06 Precipitator Wash Schedule - The precipitator was washed at the completion of the type "B" testing. Cycles with and without detergent addition were run resulting in the recommendation by representatives of the manufacturer that detergent addition be included as part of the wash cycle. Observation of the collecting plates after washing indicated that they remained discolored (black) but no excessive build-up of

unremoved particulate.

On the basis of experience during the initial testing phase, a schedule of one wash per week was established as a trial procedure for the interim phase of the test program where the prototype would be operated during normal production testing of engines. Unfortunately, the schedule was not rigorously followed during the interim period and thus no reliable data was obtained relative to the adequacy or inadequacy of washing.

In the absence of field data, the precipitator manufacturer was consulted regarding his experience with similar installations. It is their estimate that with particulate grain loadings in the range experienced during testing, the first collecting field would require washing once per 40 hours of operation and the second field once per 160 hours of operation. This information must be regarded as approximate, however, due to the uniqueness of the application.

The frequency with which the precipitator must be washed is heavily dependent on the emission factors of the engines tested. Since these factors can vary widely, washing schedules will vary widely depending on the engines tested.

SECTION 5

FULL-SCALE SYSTEM DESCRIPTION AND ECONOMICS

5.01 System Description - Design of a full-scale pollution abatement system for a particular cell would be tailored to the requirements of the engines tested. Basic system components are illustrated in Figure 5.1.

5.01.1 Precipitator - The precipitator has five major components: entrained moisture eliminator, fields, washer assemblies, power control cabinets (power packs) and washer control cabinets. The moisture eliminator is of the vertical multi-pass louver design extending across the entire face of the precipitator. The louver will also serve to distribute the flow of incoming exhaust gas. The precipitator fields are comprised of a number of individual cells stacked in vertical (modules) banks perpendicular to the direction of gas flow and supported in a structural frame. Each cell has dimensions of 36" wide, 14" deep and 16" or 20" high. Modules can be configured to suit the stack dimensions of a particular application. Each horizontal row of cells receives an independent power supply (two leads - 12kV and 6kV) which is connected to the outside cell in each row. Interior cells are energized by means of contact strips attached to each cell. The power to each cell is controlled by remotely mounted power packs which consist of transformer, rectifiers, voltage control and circuit Input to the control cabinets is 120V AC single phase and protection. output is 12,000V DC (ionizing fields) and 5800V DC (collecting fields).

The modules containing the precipitator fields will be arranged two in series in the direction of gas flow. Washer assemblies are located on the inlet side of each module. Each washer assembly is approximately 4'-0" wide and extends the full height of the module. During

the wash cycle, three rotating spray nozzles make four vertical passes over the entire height of a four foot wide section of the module. After four passes, the assembly indexes along a horizontal track to the next 4'-0" section and repeats the process until one complete horizontal pass is made. The nozzles travel at a speed of 6 ft/min in the vertical direction and consume 15 GPM while in operation. Cleaning detergent will be added at a rate of 1/2 GPM during the first two vertical passes via a separate pump. Washer assembly travel, water supply and detergent supply are automatically controlled from a remotely mounted control cabinet.

All components in contact with the exhaust stream are enclosed in a gas tight casing top, bottom and sides. Access doors are provided on each side of the washer assembly cavity for inspection and maintenance.

Figure 5.2 illustrates the general arrangement of components.

Each 25' X 12' section would be furnished with individual inlet plenum

and stack and appropriate turning vanes for gas distribution as determined by
a pre-design model study.

- 5.01.2 Evaporative Cooling System Test cell exhaust gases would be cooled in a manner similar to the cooling systems now in service in cells which test after-burning engines. Components include a series of spray rings located in the augmentor tube and the base of the test cell stack, control valves for modulation of spray water supply to minimize over spray, spray water pumps and water storage tank.
- 5.01.3 <u>Water Reclamation System</u> Excess spray water and the drains from the precipitator wash cycle flow by gravity to a collecting sump where they are pumped to a holding tank designed to provide surge capacity for the relatively large amount of overspray during the after

burner tests. From the holding tank the slurry is pumped through a pressureleaf filter where suspended particulates are removed and then back to the storage tank for reuse. Water from the wash cycle, one half of which would contain a biodegradable detergent, would be discarded.

- 5.01.4 Solids Removal and Recovery Solids are removed from the overspray and wash water by direct filtration. This is a relatively expensive method of solids removal but due to its compactness is suited to the ground space limitations around existing cells. The filter medium is a series of vertical hollow leaves coated with a filter aid. A mixture of particulates and filter aid are deposited on the outside surface of the leaves forming a cake. Collected particulates are removed on an intermittent basis by evacuating the filter, air dyring and vibrating the leaves which deposit the dry cake on a continuous conveyor for collection.
- 5.01.5 System Controls The system would operate automatically during the engine test. Operator action would be required in the following areas:
 - Turn on power to equipment prior to engine test.
 - Remotely monitor (annunciator) systems during test.
 - Initiate and monitor precipitator wash cycle.
 - Initiate and monitor collected particulate removal from pressure leaf filter and returning of filter to service.

A typical arrangement of system components is illustrated in Figure 5.3.

5.02 <u>Capital Costs</u> - The cost of installing a two-stage precipitator,
evaporative cooling system and sludge removal system capable of handling

550,000 ACFM of test cell exhaust gas is estimated to be \$850,000.

This size installation would be large enough to handle a J-79 (180 lb/sec)

engine with after burner or a TF30 (250 lb/sec) engine without after burner.

Estimated cost of a larger system capable of handling the 1,200,000 ACFM of exhaust which would result during test of a 350 lb/sec turbo fan engine in after burner is \$1,690,000. Cost breakdowns are shown in Table 5.1.

The basic precipitator for the 550,000 ACFM unit would consist of eight (8) modules each 24'-8" wide by 12'-0" high. On a test cell with the stack configuration of those at Black Point, these modules would be arranged one wide, two high and two deep (in direction of gas flow) on two sides of the stack. The 1,200,000 ACFM unit would require sixteen (16) modules 24'-8" X 13'-0" arranged two high and two deep on all four sides of the stack. The foregoing represent two arrangements of precipitator surface which appear workable. Module dimensions and physical arrangement can be varied to suit stack configuration and ground space availability at particular cells.

5.03 Operating Costs - Estimated operating costs for the pollution abatement system are summarized below for the areas of consumable utilities, consumable material, maintenance and operating labor.

5.03.1 Consumable Utilities - The overall cost of utilities will vary considerably with the size and type of engine tested and the test duration due to the large cost impact of the evaporative cooling system.

Table 5.2 summarizes estimated utilities consumption and cost for a J-79 engine with after burner representative of the 500,000. ACFM cell and a hypothetical 350 lb/sec engine with after burner representative of the 1,200,000 ACFM cell. The basis for quantities listed in Table 5.2 is given in Appendix 3.

An additional utility would be compressed air for drying the pressure filter prior to cleaning. This would amount to approximately 10,000 SCF per cleaning: approximate cost = \$2.00.

TABLE 5.1 CAPITAL COST OF TEST CELL POLLUTION ABATEMENT SYSTEMS

	550,000 ACFM	1,200,000 ACFM
Precipitator assemblies incl. cells, washers, moisture eliminator, controls, support frame, casing, breeching and stacks.	\$474,2 00	\$1,112,600
Evaporative cooling system incl. water storage tank, spray pumps, piping, nozzles and controls.	\$ 71,900	\$ 133,000
Water reclamation system incl. sump pump, surge tank, slurry pumps, piping and controls.	\$ 26,500	\$ 40,200
Pressure-leaf filter assy incl. filter, supports, precoat tank and piping.	\$ 65,800	\$ 65,800
Electrical work incl. power transformers, motor controls, lighting and wiring.	\$ 92,600	\$ 134,400
Civil/structural work incl. site work, support steel, walkways, equipment slab and control building.	\$118,900	\$ 204,400
TOTAL CAPITAL COST	\$849,900	\$1,690,400

Prices include markups as follows: Omissions and Contingencies - 15% on all items except precipitator and pressure filters.

Contractors OH&P - 21% on all items.

General Contractors OH&P - 5% on structural and electrical totals.

TABLE 5.2
UTILITIES CONSUMPTION AND COST PER ENGINE TEST

		J-79 w/A	B	350 lb/sec w/AB		
	Source	Consumption	Cost	Consumption	Cost	
POWER @ \$0.3/ KWH	Precipitator Energization Evaporative	5 6 KWH 66 KWH	\$1.68 \$1.98	146 KWH 143 KWH	\$4.38 \$4.29	
	Cooling Pumps Spray Reclamation Pumps	2.3 KWH	\$0.07	4.3 KWH	\$0.13	
	Precipitator Washer* TOTAL POWER	0.3 KWH 124.6 KWH	\$0.01 \$3.74	0.9 KWH 294.2 KWH	\$0.03 \$8.83	
WATER @ \$.35/ 1000 Gal.	Lost Through Evaporation Discarded wash Water**	28,800 Gal. 363 Gal.	\$10.08 \$ 0.13	64,500 Gal. 856 Gal.	\$22.58 \$ 0.30	
	TOTAL WATER	29,163 Gal	\$10.21	65,356 Gal.	\$22.88	

*Precip Washing Power: J79 - 4.6 KWH/Wash 16 Tests/Wash

350 lb/sec - 14.1 KWH/Wash 16 **T**ests/Wash

**Discarded washwater: J79 - 5800 Gal/Wash 16 Tests/Wash

350 lb/sec - <u>13700 Gal/Wash</u> 16 Tests/Wash 5.03.2 <u>Consumable Material</u> - Recurring consumable items include detergent for the precipitator wash cycle and filter aid for the pressure filter system.

The material required for the filter aid will fluctuate widely depending upon the amount of particulates filtered and the ratio of filter aid to particulate which must be maintained to provide a porous cake. The amount of particulates collected will be proportional to the emission factors (e.g. dirtiness) of the engines tested. These values apparently can range from the 1.0-1.7 lbs/10⁶BTU reported by previous air pollution tests (1) to the 0.16-0.24 lb/10⁶BTU experienced in the Jacksonville tests. Ratio of filter aid to particulate required can range from 0.1 to 1.0 lb F.A./lb particulate. This ratio is empirically derived for a particular installation.

A pressure leaf filter with 800 sq.ft. of filter surface could collect approximately 1400 lbs of particulates between cleanings. Using a 1:1 ratio of filter aid to particulate and 0.1 lbs/sq. ft. from the precoat cycle, a total of 1480 lbs of filter aid would be required for a complete cycle. The cost per cycle would be approximately \$44.40 using a \$3.00/100 lbs material cost based on an east coast location.

Using conservative emission factors in the 1.0-1.7 lbs/
10⁶ BTU range, it is estimated that a J-79 in a 120 minute cycle would
generate 212 lbs of collected particulates and a 350 lbs/sec turbofan in
a 137 minute cycle would generate 291 lbs of particulate. This rate of
collection would require filter cleaning twice every 12-13 tests with the
J-79 or 9-10 tests with the 350 lb/sec engine.

Detergent is supplied at a rate of one part to 40 parts water during one-half of the wash cycle (two of the four passes). Consumption

for one complete wash cycle is therefore:

500,000 ACFM unit: 5,760 Gal X $\frac{1}{2}$ cycle X 1/40 ratio = 72 Gal.

1,200,000 ACFM unit: 12,500 Gal X ½ cycle X 1/40 ratio = 156 Gal.

Using a material cost of \$4.00/gallon, the cost per wash is \$290 for the

500,000 ACFM unit and \$629 for the 1,200,000 ACFM unit. As previously mentioned,

washing schedule can vary considerably, therefore, for estimating purposes

only it will be assumed that on both systems the first bank is washed every

10 tests and the second bank every 40 tests. This averages to one complete

wash every 16 tests for a cost of \$18.14 per test for the 500,000 ACFM unit and

\$39.30 for the 1,200,000 ACFM unit.

- 5.03.3 <u>Maintenance Costs</u> Elements of the systems which could be expected to require periodic replacement are as follows:
 - Precipitator (per modulo per year)
 - Fields Ionizer wire replacement; 100 per year @ \$1.70 ea = \$170
 - Cell replacement; 3 per year @ \$275 ea = \$825
 - Power Packs Transformer replacement; one every four years @ \$236 ea = \$59
 - Silicon rectifier replacement; one per year @ \$26 = \$26
 - Washer Assy Replacement of motors and chain drives after 15 years = \$200/15 = \$13
 - Washer Control Miscellaneous component replacement over 15 year life = \$1000/15 = \$67/year

Total estimated annual material replacement costs per module = \$1160. Cost for 8-module 500,000 ACFM installation = \$9280. Cost for 16-module 1,200,000 ACFM unit = \$18,560.

- Pump Maintenance: Lubrications, seals, etc. = \$200/year
- Pressure Filter: Main gasket replacement one per year = \$35

- Controls and instrumentation: Estimate of \$200/year
In addition to parts replacement, the installation would
require routine inspection on a weekly basis. At intervals of approximately
5 years, all cells should be removed and manually washed. It is estimated
that this operation would require approximately 4 mandays per module.
Allowing one manday per week for routine maintenance, average annual
maintenance labor requirements for the two systems investigated would be
as follows:

500,000 ACFM: Routine maintenance: 1 md/wk X 52 weeks = 52 md/yr

Major cleaning: 2 md/wk X 8 modules/5 yrs = 32 md/yr

TOTAL: 55.2 md/yr

1,200,000 ACFM: 2 md/wk X 52 wks = 104 md/yr

2 md/module X 18 modules/5 years = 6.4 md/yr

TOTAL: 110.4 md/yr

At an average cost of \$75/md, the annual costs for the two systems would be \$4140 and \$8280 respectively.

5.03.4 Operating Labor - The system is designed to operate automatically with operator action required only to turn on the power supply to the precipitator. An annunciator panel would be provided in the test cell control room to alert operators to any off-normal conditions which may arise. Precipitator washing and removal of carbon sludge from the filter is also highly automated requiring operator action only to initiate the cycles and monitor the precoat operation. In view of the above, we do not anticipate the need for any additional full time staffing to operate the system. A portion of the operators! time would have to be devoted to

supervision of the wash cycle and solids separation processes. This would be in the range of 3 hours per wash/removal cycle assuming full time supervision while the equipment is operating.

5.03.5 Annual Operating Costs - Annual costs are summarized in Table 5.3 for an assumed test cell loading of 500 engine tests per year. Approximately 34% of the above costs represent consumable material used in precipitator washing and solids separation whose useage is basically a function of engine dirtiness.

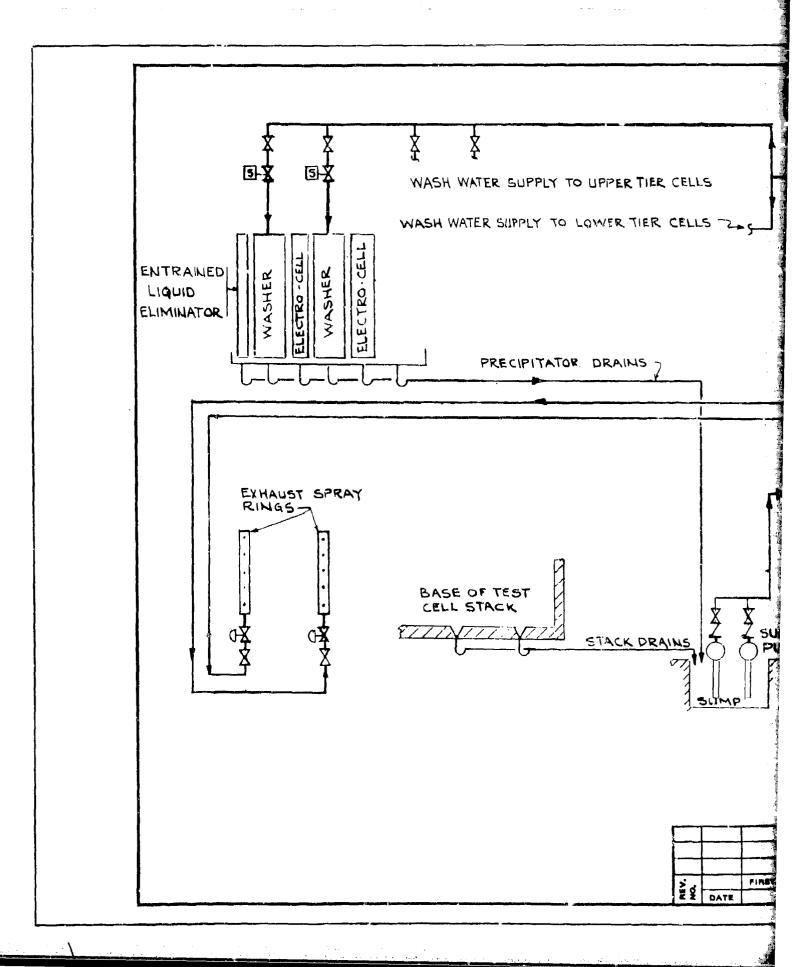
TABLE 5.3

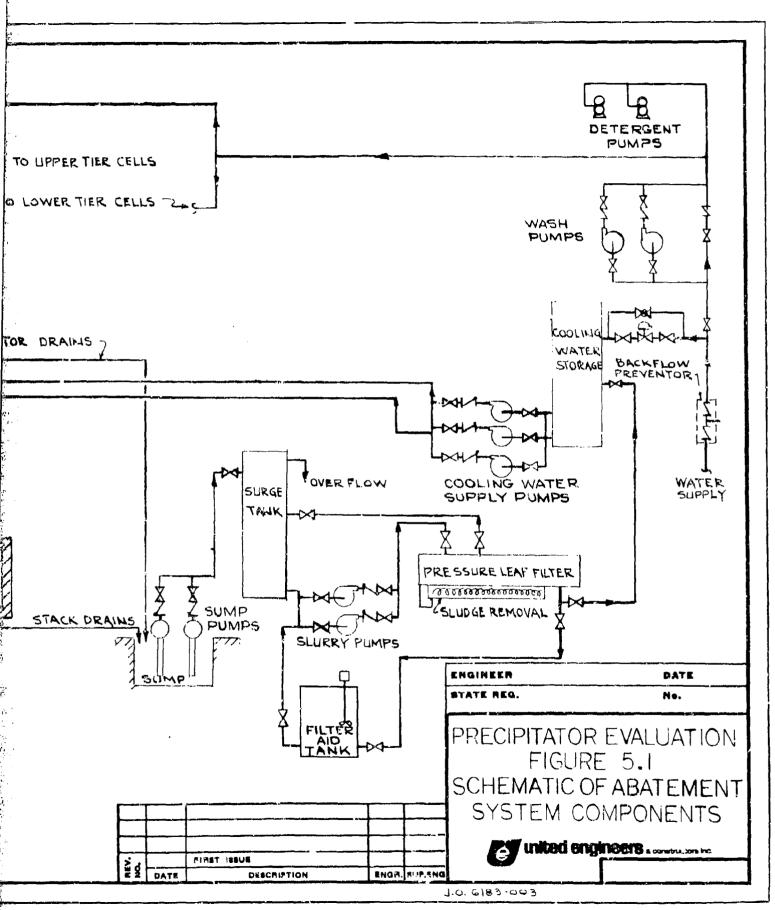
ANNUAL OPERATING COSTS

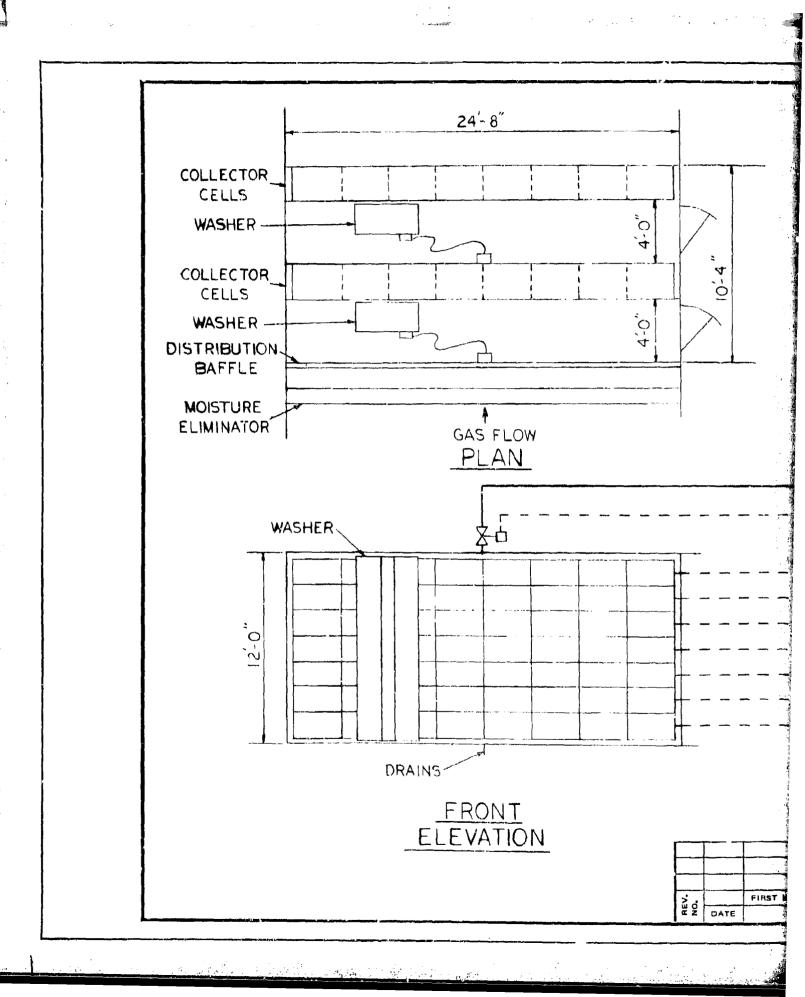
500 ENGINE TESTS PER YEAR

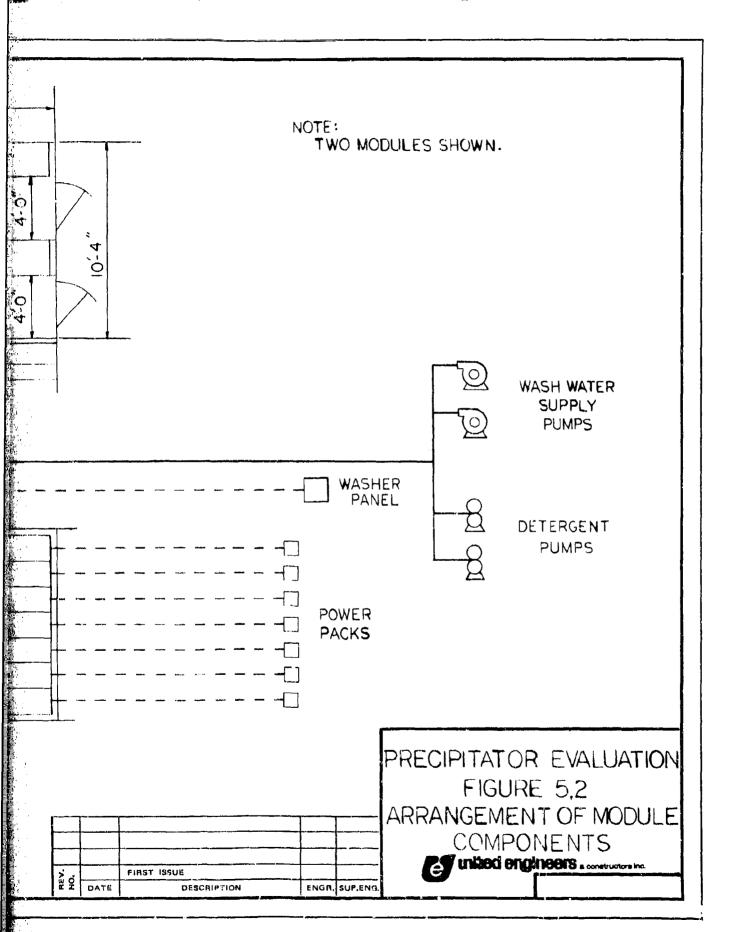
				Γ**	
	It	em	Consumption	Rate	Annual Cost
	Utilities	- Power*	62,300 KWH	\$0.03/ KWH	\$1,869
		- Water	14.58 X 10 ⁶ GA	\$0.35/ 1000 G.	\$5,103
ACFM)	Material	- Detergent	2,250 Gal.	\$4.00/ Gal.	\$9,000
8		- Filter aid	61,667 lbs.	\$3.00/ 100 lbs	\$1,850
(500,000	Maintenance	- Parts			\$9,715
		- Labor			\$4,140
J-79	Totals	3			
					\$31,667
ACFM:)	Utilities	- Power*	147,100 KWH	\$0.03/ KWH	\$ 4,413
8		- Water	32.68 X 10 ⁶ Gal.	\$0.35/ LOOO G.	\$11,438
(1,200,000	Material	- Detergent	4875 Gal.	\$4.00/ Gal.	\$19,500
1 1		- Filter Aid	74,000 lbs.	\$3.00/ 100 1b	\$ 2,280
lb/sec	Maintenance	- Parts			\$18,935
		- Labor			\$ 8,280
350	Totals	3			ውሬ <i>ት ወነራ</i>
				m./	\$64,846

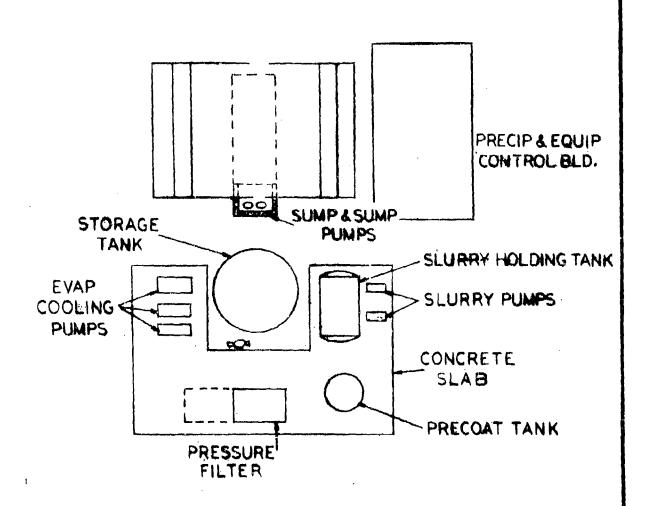
^{*} Exclusive of demand charges











PRECIPITATOR EVALUATION
FIGURE 5.3
GENERAL EQUIPMENT
ARRANGEMENT

SECTION 6

APPENDIX

- A-1 Performance Evaluation Conducted on a Two-Stage Electro-Cell Unit, Jet Engine Test Cell, by American Air Filter Co. Inc. 23 May 1975 and 24 July 1975.
- A-2 I-O Memorandum: Air Samples from Electrostatic Precipitator; Results of; w/enclosures, Naval Air Rework Facility, NAS Jax 16 June 1975.
- A-3 Basis for Operating Cost Computations
 - Parameters for typical engine test cycles
 - ~ Precipitator energization
 - Evaporative Cooling
 - Spray Reclamation
 - Precipitator Washing
 - Pressure Filter System
- A-4 Conversion of test data to emission factors
- A-5 Dimensional Drawing of Prototype Two-Stage Precipitator

APPENDIX A-1

Performance Evaluation Conducted on a

Two-Stage Electro-Cell Unit

Jet Engine Test Cell

American Air Filter Co., Inc.

TEST REPORT
Pilot Two Stage Precipitator
United Engineers and Constructors Inc.
Purchase Order No. BOS-287
NAVFAC Contract N62467-74-C-0161
NARF Jacksonville, Florida

CONTENTS

- 1. Report No. 1, 23 May 1975
- 2. Report No. 2, 24 July 1975
- 3. Drawing No. DEV-835A

PEP #702 23 May 1975

Performance Evaluation
Conducted on a Two-Stage Electro-Cell Unit
Jet Engine Test Cell
Naval Air Station
Naval Air Rework Facility
Jacksonville, Florida

SCOPE:

During the weeks of April 6 and April 13, 1975, a field trip was made to the Naval Air Station located in Jacksonville, Florida. The purpose of this trip was to accomplish performance testing on a two-stage Electro-Cell unit installed on the exhaust system of a jet test cell facility. The tests were conducted in order to establish the unit's efficiency at various flow rates. Based on the results of these tests, an optimum flow rate, which would provide approximately 90% efficiency on the exhaust fume, may be determined.

BACKGROUND:

In operation, the jet test cell is utilized as a permanent stand for testing rebuilt and repaired jet engines, the exhaust of which is emitted to the atmosphere after passing a wet scrubber unit. In search of a better cleaning device, the double unit Electro-Cell test system was installed at this site and positioned such that a portion of the exhaust could be drawn from the main scrubber stack prior to actual scrubber entry (see Illustration A).

ZLECTRO-CELL SETUP & PREPARATION:

The Electro-Cell test unit consisted of two (2) ECU-5 units with washers, arranged in series (no fan). An SG-7 power pack was used for each ECU. Ionizer voltage on both packs was set at 13.5 kV. Plate voltage was 6.3 kV. Current draw to the two packs during tests was extremely high due to excessive arcing (see below). Readings would fluctuate from 4 to 8 amps for both packs (2 to 4 amps each). Each washer was controlled by a separate standard arrangement II washer control with detergent option.

The test duct originated with a one foot by one foot duct inlet, located about 10 feet downstream and directly in line with the J-79 jet engine exhaust and cooling water spray rings. Air flow through the unit was regulated by use of various size slocted plates over this inlet.

Initial tests using a damper located downstream of the ECU's for air flow adjustment resulted in severe damage to the rather old test duct. This was due to the tremendous thrust (or velocity pressure) from the engine which built up in the scrubber base section where the sample duct inlet was located.

In addition, extreme arcing in the Electro-Cell elements caused reliability problems with the power packs. This problem was a result of the extremely high entrained moisture content and saturated condition of the test air. Water flow from the sump of the test duct upstream of the test unit was in the magnitude of 15 liters per minute from about 7000 CFM of air. Approximate water flow from the ECU unit drain was 1 liter per minute.

The only deviation from the standard ECU was in the use of milicone insulated high voltage wiring from the power packs to the buss bars. Insulation on the standard wiring was burned off at the buss bar end due to the wet, dirty condition causing are paths along the outside of the insulated wire to ground.

The location of the test duct inlet, directly in the jet engine exhaust stream, will require, for reliable operation, some sort of weather louvre or other entrained water eliminator arrangement. Permanent installations must be designed to eliminate entrained moisture for reliable operation of the Electro-Cell.

REQUIRED TESTS & EQUIPMENT:

Tests to be performed by AAF personnel were:

- A. Determine air flow volumes at various pressure drops across the venturi section, installed upstream of the Electro-Cell banks.
 - B. Determine the Electro-Cell efficiencies at various flow rates.

Regulation and measurement of air volumes were accomplished by utilization of inlet blank-off plates and pitot tube traverses in the venturi inlet duct rather than using the venturi pressure drops. This method provided a more precise means of determining the actual flow since the venturi pressure "bounced" continuously, creating a problem of accurately depicting the true pressure drop.

Efficiency evaluations were obtained using inlet and outlet dust grain loads determined from AAF five-inch dust sampling equipment. These samplers are designed for capture and measurement of the particulate content in a gas stream and are not intended for any gas analysis. The unit consists of a probe of sufficient length on which various size tips can be installed. The probe in turn is attached to the sampler header which houses the filtering media and monitoring orifice. This header is enclosed

in a Glass-Col heating mantel with appropriate temperature monitors and controls. The header is then attached to a sufficient vacuum source to provide air movement. By correlation of probe tip sizes and orifice pressure regulation iso-kinetic sampling conditions can be established and maintained.

Filtering media employed for these tests is defined as H-93 super-fine glass with initial capability of 99.97% retention of 0.3 micron particles. Retention ability rises as media load increases.

TEST PARAMETERS & RESULTS:

A. Air Flow

Establishment of system sir flows was an essential portion of the test program, since sampling rates and operational functions of dust sampling equipment is directly related to these flows. The initial intent was to utilize a venturi (calibrated) for air flow measurement, however, as previously indicated, this approach failed to provide the desired degree of accuracy. The method employed consisted of utilizing four sizes of inlet blank-off plates at the entrance duct to the Electro-Cell unit. Approximately six feet downstream of the inlet and immediately prior to venturi entry, a pitot tube traverse was conducted for each of the four inlet plates. Each traverse consisted of sixteen (16) check points taken in the center of three-inch squares, having divided the one square foot duct into sixteen individual squares. The velocity pressures at each point were obtained and recorded along with temperature and barometric pressures. From these traverses, the total air volumes were calculated as indicated in Table #2, followed by actual calculation and computer data. The results of these traverses were as follows:

- 1) 12 square inch opening inlet plate provided 2517.6 actual cubic feet per minute.
- 2) 18 square inch opening inlet plate provided 3827.6 actual cubic feet per minute.
- 3) 24 square inch opening inlet plate provided 4732.0 actual cubic feet per minute.
- 4) 36 square inch opening inlet plate provided 7334.1 actual cubic feet per minute.

Using then, an entry area to the Electro-Cell unit of 14.63 square feet, these values provide cell velocities of 172.1, 261.6, 323.4, and 501.3 feet per minute, respectively.

B. Dust Sampling & Efficiencies

Having established total system flows, a series of upstream and downstream dust loading samples were obtained using the AAF 5-inch sampling devices. Since velocity pressures are incorporated in the formulation of iso-kinetic sampling rates, it was necessary to derive the velocity pressure by calculation rather than direct measurement due to the low pressure values. From Table 1, it will be noted that duct flow in feet per minute is indicated for each of the four inlet plates employed. (Duct area was 10 square feet.)

Using these volumes, we are able to calculate the average velocity pressure in the duct by means of

$$V_p = \frac{\text{Velocity (FPM)}}{4005}$$

where V_p is the velocity pressure, v=velocity in feet per minute, and 4005 the constant. Having derived the velocity pressure values, sampling rates were determined for upstream and downstream units incorporating various sized inlet tips. However, since the velocity pressure values are the average at the particular system volumes, it was determined that sampling rates slightly larger than iso-kinetic should be used during sampler runs to insure that sufficient sampling was accomplished. This will not affect the results of the loading tests since each cubic foot of air contains a volume of dust consisting basically of 0 - 5 micron material which has negligible gravitational and inertial forces acting upon it, due to the minute size. This volume vs. air sampled provides the actual dust loading in grains per cubic foot.

The individual tests, consisting of upstream and downstream sampling, conducted simultaneously, were performed at various unit flow as illustrated in Table 1. As will be noted from this Table, the inlet grain loadings ranged from 0.00120 to 0.00290 grains per cubic foot of air and the outlet grain loadings varied from 0.0005 to 0.0001 grains per cubic foot. The efficiency band with both Electro-Cell units energized varied from 82.75% to 97.64%.

Compiling the results of the total test program, it was established that the actual flow rate at which the two-stage Electro-Cell would provide approximately 90% efficiency was 501 FPM cell velocity with the total system flow at 7334.0 CFM. Also one test was conducted with only one bank of Electro-Cells energized and the test conducted at these conditions provided an 80% collection efficiency.

A summary of all tests is illustrated in Tables 1, 2, and 3, followed by actual test data and computer printouts.

Richard P. Williams

Wilson Welch

Wilson Welch

ws: 19 MAY 1975

ATTACHMENTS

cc: PEP#662
K. Westlin
H. McShane
J. Ashe
Orig. - PEP #702

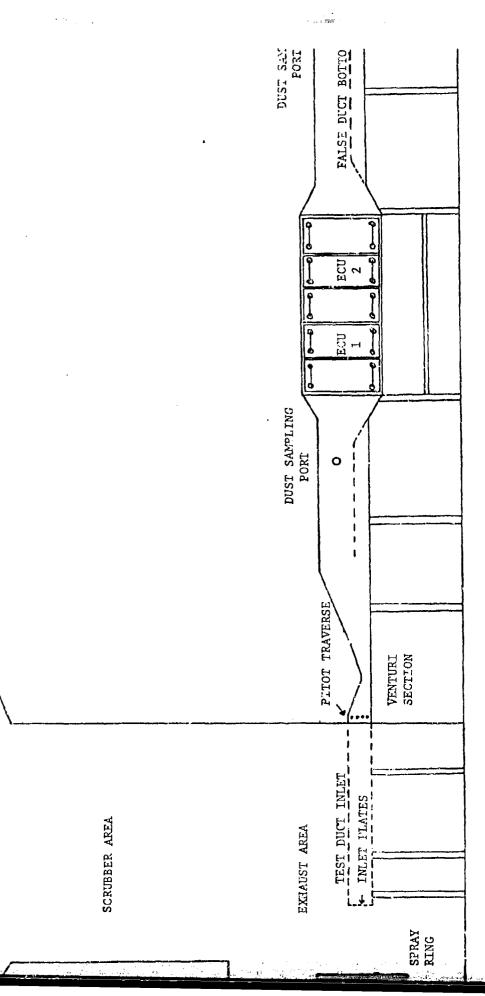


ILLUSTRATION A. Test System Installation (Not to Scale)

TABLE 1 OPERATIONAL PARAMETERS & TEST RESULTS SUMMARY* FOR AAF TWIN ELECTRO-CELL UNIT AND TEST ENGINE J-79-10, SERIAL #433544

4/15/75	P-8(1)	15:15 15:15	36	7050	7319	5877	Ĺ76	73	128-132	29.98	7334.1	733.4	501.3	,0015356	0.0003058	80.08
4/14/75	1-4	15:00	36	9569	7138	7132	987	76	131-133	29.58	7334.1	733.4	501.3	0.00270	0.0005	80 80 80
4/14/75	P-6	11:63	36	7003	7373	6071	987	71.5	131-132	30.13	7334.1	733.4	501.3	0.00270	0.0002163	91.98
. 4/11/75	7-5	17:00	24	8669	7160	5924	985	7.5	131-133	29.77	4732.0	473.2	323.4	.00192	.00010	94.79
4/11/75	.7	14:40 16:10	54	Ü001	7325	0909	866	ĸ	131-133	29.75	4732.0	473.2	323.4	0.00194	C. 00001465	97.64
4/10/75	γ-A	15:57	24	6974	7040	5831	975	74	131-133	29.88	4732.0	473.2	323.4	•	•	ı
4/10/75	P-3	11:30	18	7000	7345	909	056	8	128-132	29.95	3827.6	382.7	261.6	0.00121	0.0000401	96.68
4/10/75	V-3	10:15	18	7000	2400	6194	7007	7.1	131-133	29.98	3827.6	382.7	261.6	,	1	•
4/9/75	P- 2	13:55	12	7000	7220	5983	988	73-73	130-133	30.05	2517.6	251.7	172.1	0.00120	0.0001465	87.79
51/6/7	7	08:40 09.32	12	7500	7500	0819	542	11-89	130-132	30.10	2517.6	232.7	172.1	0.00290	c. noos	82.75
4/8/75	4 ~2	12:25	71	7000	7340	7709	686	73	127	30.06	2517.6	251.7	172.1	•	•	•
4/1/75	V-1	62:30	36	7000	731.0	8709	766	ĸ	131	30.07	7334.1	733.4	501.4	1	ı	•
DATE OF TEST	TEST NUMBER	ACTUAL IEST TIME	ORIFICE SIZE (IN ²)	TNGINE RPH (AVG)	THRUST (AVG)	FUEL FLC4 (AVG)	E.G.I. (AVG)	AYBIENT TEMPERATURE F	INLET DUCT TEMPERATURE F	BARCHETRIC Pressure IRHz	TOTAL AIR VOLUNE (IN GPM)	MCT VELOCITY (AR MI)	CELL VELUCITY (IN FPM)	INLET CONCENTRATION	CUTLET CONCENTRATION	SYSTEM EFFICIENCY, 7

(1)System operating with only one set of Electro-Cells energized. *Actual data and computer prints given in Section A.

TABLE 2
Pitot Tube Velocity Traverse Summary
on Venturi Inlet

DATE .	4/7/75	4/8/75	4/10/75	4/11/75				
TEST NO.	V-1	V-2	v−3	V~4				
ORIFICE SIZE (IN ²)	36	12	18	24				
BASEMETRIC PRESSURE ABSOLUTE	30.07	30.07	29.98	29.88				
DRY BULB TEMPERATURE	130.0	127.0	130.0	130.0				
WET BULB TEMPERATURE	130.0	127.0	130.0	130.0				
DENSITY - 1bs/ft ³	0.06377	0.06440	0.06357	0.06334				
PITOT TUBE CORR. FACTOR	1.0	1.0	1.0	1.0				
INLET SIZE (FT ²)	1.0	1.0	1.0	1.0				
NO. OF POINTS	16	1.6	16	16				
ACFM*	7334.1	2517.6	3827.6	4732.0				
SCFM*	6236.1	2161.8	3244.3	3996.6				

^{*}Calculation formulas described on following page.

INCH SAMPLE TEST RESULTS

	QSSUM QDASUM CONC/ACF CONC/DGSCF		100.8518 64.8744 84.4977 0.0029 0.0045 0.0035		115.2485 74.9531 97.8162 0.0012 0.0018 0.001463 180.3103 118.6034 154.2770 0.0001 0.0002 0.0002	111.4972 147.4464	176,1802 235,4408 0.0000401 0.0001	274.9590	330.2221 444.2592 .0000436 0.0031	276.5499	514,0999 328,3407 441,8598 0.0001 U.0002	0.00270	506.4850 320.8162 429.1876 0.0002165 0.0003	211 0631 134.9075 180.0978 0.00270 0.0043	213.9625 287.7871 0.0003 0.0004	135 935 180 2832 0.0015356 0.0023	
10	Pad Gain Test Duration	Grams Minutes	0.0189	0.0041	0.0086	77.00	0.0007	70,07	0.0015	0070	0.0032		0.0543		0.0376 0.0062		0.0208
		rad No.	٧-839		CI-518		CT-521 CT-520	מון מון	CT-523 CT-522	;	CI-525 CT-524	1	CT~531	3	CT-529	1	CI-527
	Test No.	& Position	1_Inctream	1-Upstream 1-Downstream	2-Upstream	2-Downstream	3-Upstream	3-Downstream	4-Upstream	4-DOWIISCE COM	5-Upstream	3-DOWES LI CAM	6-Upstream	6-DOWNSLIES	7-Upstream	/-Downsrream	8-Thetream
		Date	•	6/16/4	4/9/75		4/10/75		4/11/75		4/11/15		4/14/75		4/14/75		2/15/75

QSSUM = total actual sampled volume at sampler temp. and press., ACF. QDASUM = total actual scupled volume at duct temperature and pressure. QDSUM = total dry gas sampled volume at 70 degrees F and 29.92 in HG.

CONC/ACF = grains per cubic foot at sampler conditions, gr/ACF.

- grains per cubic foot of dry gas at 70 degrees F and 29.92 in HG. CONC/DGSCF

CONC/DUCT ACF = grains per cubic foot at duct conditions, gr/duct ACF.

Calculation of Air Volume from Pitot Traverse Computer Program 7058

Air Calculations

(gas option = 1)

DENS = $\frac{P - .38 (PSAT - P(DB - WB)/2700}{.754 (DB + 459.6)}$ (From AMCA Standard Test Code Bulletin 210 Section TV)

P = Duct Pressure in in. Hg = PB - PD

PB = Barometric Pressure in in Hg

PD = Duct Pressure depression in in. Hg

DB.WB = Duct Temperatures in OF

PSAT = Saturation Temperature at WB in in.Hg

ACFM = 1096.5 A $(\frac{\Sigma \sqrt{VP}}{N \sqrt{DENS}})$

A = Duct Area in ft²

VP = Corrected pitot velocity pressure in in. WG

N = Number of traverse points of VP

SCFM = ACFM $(\frac{DENS}{.075})$

DGCFM = ACFM $(\frac{70 + 459.67}{DB + 459.67})(\frac{P - PV}{29.92})$

 $PV = (PSAT' - \frac{.3895 P' (DB - WB)}{1093.8 - .576 x WE})(2.036)$ in in Hg

PSAT' and P' in psia

```
FORM 07058 MAYEAC INLET TO VENTURI HIR VOLUME TEST 1 7APR75
                                                                     PAGE
 04721775
               10:19
    DUCT AREA CODE = 3
                                      CORRECTION CODE = 1
    GAS CODE
                                      BARDMETER ABS = 30.07000
    DRY BULB TEMP = 130.00000
                                      WET BULB TEMP = 130.00000
                                      WIDTH
                                                     = 12.00000
                   = 12.00000
    LENGTH
    PITOT CORRECTION FACTOR = 1.00000
    NO. OF POINTS = 16
           VELOCITY PRESSURE
POSITION
                2.500
    1
                2.500
    3
                2,750
                2.750
    4
    5
                3.000
                3.000
                3.000
                3.000
    8
    9
                3.000
                3.000
   10
                2.900
   11
                3.100
   12
                2.600
   13
                2.700
   14
                2.900
   15
                3.000
   16
```

RM = 1.6890898 DAREA = 1.000 SQ FT
DENS = 9.06377 RCFM = 7334.1

SCFM = 6236.1

```
FORM 07058 NAVEAC INLET TO VENTURI AIR VOLUME TEST 2 SAPR75
                                                                       PAGE
                  10:34
 04/21/75
                                       CORRECTION CODE = 1
    DUCT AREA CODE = 3
                                       BARDMETER ABS = 30.07000
    GAS CODE
                                       WET BULB TEMP = 127.00000
    DRY BULB TEMP = 127.00000
                                                      ·≈ 12.00000
                                       WIDTH
                   = 12.00000
    LENGTH
    PITOT CORRECTION FACTOR = 1.00000
    NO. OF POINTS = 16
MOITIZO
           VELOCITY PRESSURE
                 0.240
   11
                 0.320
    12
                 0.300
    13
                 0.280
    14
                 0.320
    21
                 0.450
    23
                 0.350
    23
                 0.340
    24
                 0.350
    31
                 0.350
    32
                 0.350
    33
                 0.360
    34
                 0.300
    41
                 0.350
   . 42
                 0.400
    43
                 0.400
    44
                                                              1.000 SQ FT
                                        DAREA
                      = 0.5826707
     RM
                                                              2517.6
                                        ACFM
                          0.06440
     DENS
```

2161.8

SCFM

```
FORM 07058 NAVEAC INLET TO VENTURI AIR VOLUME TEST 3 10 APR75
                                                         PAGE 4
               10:19
04721775
   DUCT AREA CODE = 3
                               CORRECTION CODE = 1
   GAŞ CODE
                               BAROMETER ABS = 29.98000
   DRY BULB TEMP = 130.00000
                              WET BULB TEMP = 130.00000
   LENSTH = 12.00000 WIDTH = 12.00000
```

PITOT CORRECTION FACTOR = 1.00000

NO. OF POINTS = 16

POSITION VELOCITY PRESSURE

DENS

SOFM

0.750 0.800 0.750 0.850 0.850 0.950 0.900 0.850 0.800			•
0.800 0.750 0.800 0.850 0.950 1.000 0.900 0.850 0.800			
0.800 0.750 0.800 0.850 0.950 1.000 0.900 0.850			
0.800 0.750 0.800 0.850 0.950 1.000 0.900			
0.800 0.750 0.800 0.850 0.950 1.000			•
0.800 0.750 0.800 0.850 0.950 1.000			,
0.800 0.750 0.800 0.850 0.950			,
0.800 0.750 0.800 0.850			
0.800 0.750 0.800			•
0.800 0.750			,
0.800	•		
ስ ንፍለ			
		•	
0.650			
	0.550 0.700 0.680	0.700	0.700

ACEM

0.06357

3244.3

3827.6

```
FORM 07058 NAVEAC INLET TO VENTURI AIR VOLUME TEST 4 11 APR75
                                                                   PAGE 3
 04/21/75
                 10:19
    DUCT AREA CODE = 3
                                      CORRECTION CODE # 1
    GAS CODE
                                     EAROMETER ABS = 29.88000
    DRY BULB TEMP = 130.00000
                                      WET BULB TEMP = 130.00000
                  = 12.00000
                                      WIDTH
                                                    = 12.00000
    LENGTH
    PITOT CORRECTION FACTOR = 1.00000
    NO. OF POINTS = 16
POSITION
           VELOCITY PRESSURE
   11
                1.000
                0.800
   12
                1.200
   13
   14
                1.100
                1.250
   21
   22
                1.300
   23
                1.250
   24
                1.250
   31
                1.300
   32
                1.350
   33
                1.350
   34
                1.300
   41
                1.050
                1.200
   42
   43
                1,150
   44
                1.100
    RM
                  = 1.0861518
                                      DAREA
                                                          1.000 SQ FT
    DENS
                      0.06334
                                      ACFM
                                                           4732.0
    SOFM
                         3996.6
```

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 V-839 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.00030 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 1.98190 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.01890 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.100 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 1 UPSTREAM 9 APR75 04/21/75

PAGE 2

TIME: TIMES DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH PPNH DENS 0.00 5.00 130. 130. 250. 137.9 1.67 30.100 4.526 0.0532 5.00 7.00 130. 130. 240. 1.66 137.3 30.100 4.526 0.0539 136.7 136.2 10.00 7.00 130. 230. 1.65 130. 30.100 4.526 0.0547 10.00 15.00 130. 130.. 230. 1.65 30.100 4.526 0.0547 15.00 240. 20.00 130. 130. 1.66 137.3 30.100 4.526 0.0539 20.00 25.00 130. 130. 245. 1.67 137.6 30.100 4.526 0.0536 25.00 30.00 245. 130. 130. 1.67 137.6 30.100 4.526 0.0536 130. 250. 137.9 30.00 35.00 130. 1.67 30.100 4.526 0.0532 35.00 40.00 250. 1.67 130. 130. 137.9 30.100 4.526 0.0532 40.00 45.00 130. 130. 250. 1.67 137.9 30.100 4.526 0.0532 245. 45.00 50.00 130. 130. 1.67 137.6 30.100 4.526 0.0536 50.00 52.00 130. 130. 250. 137.9 1.67 30.100 4.526 0.0532

FURM C7040 NAVEAC JET DUST SAMPLER TEST 1 UPSTREAM 9 APR75 04/21/75

PAGE 3

OBSUM = 100.8518 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF

ODSUM = 64.8744 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG

ODASUM= 84.4977 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PHD WEIGHT GAIN = 0.0189 GRAMS

100

CONC./ACF \Rightarrow 0.0029 GRAINS PER CUBIC FUOT AT SAMPLER CONDITIONS, GR./ACF $\pm 0.0036F \Rightarrow 0.0045$ GRAINS PER CU FT DF DRY GAS AT 70 DEG.F AND 29.92 IN HG

- 0 - 2000 T ACF - 0.0035 GRAINS PER CUBIC EDOT AT DUCT CONDITIONS, GREDUCT ACF

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 Y-840 PAD2

TOTAL FINAL PAD(S) WEIGHT 1.97860 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 1.97450 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.00410 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.100 IN HG

FORM C7040 HAVEAC JET DUST SAMPLER TEST 1 DOWNSTREAM 9APR75 PAGE 5 04/21/75

TIME1 TIME2 DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH PPWH DENS 2.00 128. 123. 225. 4.20 30.100 4.290 0.0553 0.00 134.8 2.00 7.00 128. 128. 225. 2.20 134.8 30.100 4.290 0.0553 10.00 225. 2.20 134.8 7.00 128. 123. 30.100 4.290 0.0553 236. 135.5 15.00 128. 10.00 123. 2.17 30.100 4.290 0.0544 255. 15.00 20.00 130. 130. 2.12 138.2 30.100 4.526 0.0528 20.00 25.00 130. 130. 245. 2.14 137.6 30.100 4.526 0.0536 245. 25.00 30.00 130. 130. 2.14 137.6 30.100 4.526 0.0536 250. 30.00 35.00 130. 130. 2.16 137.9 30.100 4.526 0.0532 240. 35.00 40.00 130. 130. 2.19 137.3 30.100 4.526 0.0539 40.00 45.00 130. 130. 245. 2.14 137.6 30.100 4.526 0.0536 137.0 235. 130. 45.00 50,00 130. 2.17 30.100 4.526 0.0543 52.00 130. 130. 235. 137.0 30.100 50.00 2.17 4.526 0.0543

FORM C7040 NAVFAC JET DUST SAMPLER TEST 1 DOWNSTREAM 9APR75 PAGE 6 04/21/75

OSSUM = 116.2136 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
ODGUM = 75.4516 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN H6
ODGGUM= 97.8994 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAR WEIGHT GAIN = 0.0041 GRAMS

MEZHOF = 0.0005 GRAINS PEP CUBIC FOOT AT SAMPLER CONDITIONS, GRZACF

SERVINGUE = 0.0008 GRAINS PER CU FT DF DRY GAS AT 70 DEG.F AND 29.92 IN HG

CONCENTRATION DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-518 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT TOTAL PAD(S) WEIGHT GAIN 2.08120 GRAMS 2.07260 GRAMS 0.00860 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.050 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 2 UPSTREAM 9 APR75 04/21/75

PAGE 2

TIME	TIMES	DUCT DB	DUCT WB	SAMPLER	DB OPD	SAMPLER I	IB PSAH	PPUH	DENS
0.00	1.00	130.	130.	250.	1.66	137.9	30.050	4.526	0.0531
1.00	3.00	130.	130.	245.	1.66	137.6	30.050	4.526	0.0535
3.00	5.00	130.	130.	240.	1.66	137.3	30.050	4.526	0.0538
5.00	8.00	130.	130.	230.	1.65	136.7	30.050	4.526	0.0546
8.00	10.00	130.	130.	235.	1:65	137.0	30.050	4.526	0.0542
10.00	15.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542
15.00	20.00	130.	130.	240.	1.66	137.3	30.050	4.526	0 . 0538
20.00	30.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542
30.00	35.00	130.	130.	230.	1.65	136.7	30.050	4.526	0.0546
35.00	40.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542
40.00	45.00	130.	130.	230.	1.65	136.7	30.050	4.526	0.0546
45.00	50.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542
50.00	55.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542
55.00	60.00	130.	130.	235.	1.65	137.0	30.050	4.526	0.0542

FOPM C7040 NAVFAC JET DUST SAMPLER TEST 2 UPSTREAM 9APR75 04/21/75

PAGE 3

Q35UM = 115.2485 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
ODSUM = 74.9531 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
ODASUM= 97.8162 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PHID MEIGHT GAIN = 0.0086 GRAMS

STATE FOR = 0.0012 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GREACE

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A ONE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

RADI CT-519 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT TOTAL PAD(S) WEIGHT GAIN 2.03620 GRAMS 2.03450 GRAMS 0.00170 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.050 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 2 DOWNSTREAM 9 APR75 PAGE 5 04/21/75

TIME1 TIME2 DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH PPWH DEHS 5.00 120. 120. 250. 4.20 130.6 30.050 0.00 3.446 0.0533 130. 235. 4.12 9.00 130. 137.0 30.050 4.526 0.0542 5.00 240. 137.3 9.00 15.00 130. 130. 4.15 30.050 4.526 0.0538 210. 3.98 15.00 20.00 130. 130. 135.5 30.050 4.526 0.0562 25.00 130. 130. 225. 4.06 136.4 30.050 4.526 0.0550 20.00 25.00 30.00 130. 130. 225. 4.05 136.4 30.050 4.526 0.0550 136.1 35,00 130. 130. 220. 4.02 30.050 30.00 4.526 0.0554 35.00 40.00 130. 130. 230. 4.09 136.7 30.050 4.526 0.0546 130. 225. 4.06 30.050 136.4 40.00 45.00 130. 4.526 0.0550 50.00 130. 130. 225. 4.06 45.00 136.4 30.050 4.526 0.0550 131. 230. 4.09 50.00 55.00 137.5 131. 30.050 4.648 0.0545 55.00 60.00 132. 132. 230. 4.09 139.3 30.050 4.773 0.0544

FORM 07040 NAVFAC JET DUST SAMPLER TEST 2 DOWNSTREAM 9 AFR75 PAGE 6 04/21/75

QSSUM = 180.3103 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS.. ACF
QDSUM = 118.6034 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 154.2770 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.
PAD WEIGHT GAIN = 0.0017 GRANS

CONC/ACF = 0.0001 GRAINS PER CUBIC FOOY AT SAMPLER CONDITIONS, GR/ACF
CONC/DGSCF = 0.0002 GHAINS PER CU FT OF DRY GAS AT 70 DEG.F AND 29.92 IN HG

THE RESIDENCE OF THE PARTY OF T

CONCENTRATION DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIF CE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-521 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT 1.98320 GRAMS TOTAL PAD(S) WEIGHT GAIN

1.99680 GRAMS 0.01360 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29,950 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 3 UPSTREAM 10APR75 04/22/75

PAGE 2

TIME1	TIME2	DUCT DB	DUCT WB	SAMPLER	מפט פט	SAMPLER (NB PSAH	PPWH	DENS
0.00	2.00	130.	130.	245.	3.75	137.6	29.950	4.526	0.0533
2.00	5.00	130.	130.	240.	3.75	137.3	29,950	4.526	0.0537
5.00	10.00	130.	130.	235.	3.75	137.0	29.950	4.526	0.0540
10.00	15,00	130.	130. •	240.	3.75	137.3	29.950	4.526	0.0537
15.00	20.00	131.	131.	245.	3.75	138.3	29,950	4.648	0. 0532
20.00	25.00	132.	132.	240.	3.75	133.8	29.950	4.773	0.0535
25.00	30.00	132.	132.	240.	3.75	133.8	29.950	4.773	0.0535
30.00	35.00	132.	132.	240.	3.75	133.8	29.950	4.773	0.0535
35.00	40. JO	132.	132.	240.	3.75	133.8	29.950	4.773	0.0535
40.00	45.00	132.	132.	245.	3.75	139.1	29.930	4.773	0.0531
45.00	50.00	132.	132.	245.	3.75	139.1	29.950	4.773	0.0531
50.00	55.00	138,	132.	245.	3.75	139.1	29.950	4.773	0.0531
55.00	60.00	135.	132.	245.	3.75	139.1	29.950	4.773	0,0531

FORM C7040 NAVFAC JET DUST SAMPLER TEST 3 UPSTREAM 10APR75 PAGE 3 04/22/75

GSSUM = 174,9916 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF QUBUM = 111.4972 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN H6 @DASUM= 147.4464 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAD WEIGHT GAIN = 0.0136 GRAMS

= 0.0012 GRAINS PER CUBIC FOOT AT SAMPLER COMDITIONS. GR/ACF CONCYACI THE CHESCE - 0.0019 GRAINS PER CU FT OF DRY GAS AT 70 DEG.F AND 89.92 IN HG

PAGE 4

i ilikalimentalian katalan kat

TEST DATA

CONCENTRATION DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-520 PAD2

TOTAL_FINAL PAD(S) WEIGHT 2.01270 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 2.01200 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.00070 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.950 IN H6

FORM C7040 NAVFAC JET DUST SAMPLER TEST 3 DOWNSTREAM 10APR75 PAGE 5 04/22/75

TIMES BUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB FSAH TIMEI DEMS 9.58 5.00 125. 125. 250. 134.1 29.950 3,955 0.0530 0.00 10.00 132, 132. 210. 9.07 137.1 29.950 4.773 U.0558 5.00 9.07 137.9 29.950 4.901 0.0557 133. 133. 10.00 15.00 210. 15.00 20.00 135. 135. 1 220. 9.17 140.1 29,950 5.166 0.0547 9.47 29.950 133.5 4.901 0,0549 20.00 25.00 133. 133. 220. 140.1 25.00 30.00 135. 135. 220. 9.17 29.950 5.166 0.9547 140.4 29.950 30.00 35,00 135. 135. 225. 9.26 5.166 0.0543 35.00 29,950 40.00 135. 135. 225. 9.26 146.4 5.166 0.0543 40.00 45.00 103. 133. 225. 9,26 103.8 29.950 4.901 0.0545 45.00 50.00 9.26 138.8 29,950 4.90) 0.0545 133. 133. 225. 29.950 50.00 55,00 133. 133. 225. 9.26 138.3 4.901 0,0545 60.00 225. 9.26 133.8 29.950 4.901 0.0345 55.00 103. 133.

FORM C7040 NAVEAC JET DUST SAMPLER TEST 3 DOWNSTREAM 10APR75 PAGE 6 04/22/75

QSSUM = 271.3594 TOTAL ACTUAL SOMPLED VOLUME AT SAMPLER TEMP. AND PRESS.; ACF
QDSUM = 176.1802 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 235.4408 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.
PAD WEIGHT GAIN = 0.0007 GRAMS
COMERANCE = 0.0000 GRAINS REP. CURIC FOOT AT SAMPLER CONDITIONS. GREACE

COMEZACE \approx 0.0000 GRAINS PER CUBIC FOOT AT SAMPLER COMDITIONS, GRZACE COMEZOGGE $\frac{1}{2}$ 0.0001 GRAINS PER CU FT OF DRY GAS AT 70 DEG.F AND 29.92 IN AG

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A ONE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PADI CT-523 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT TOTAL PAD(S) WEIGHT GAIN 2.00180 GRAMS 1.96110 GRAMS 0.04070 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.740 IN HG

FORM 07040 NAVFAC JET DUST SAMPLER TEST 4 UPSTREAM 11APR75 04/22/75

PAGE 2

TIME: TIMES DUCT DB DUCT WB SAMPLER DB DPD SAMPLER WB PSAH PPWH DENS 0.00 5.00 130. 130. 245. 5.75 29.740 137.5 4,526 0,0529 5.75 5.00 3.00 131. 131. 240. 138.0 29.740 4.648 0.0532 137.1 3.00 12.00 131. 225. 5.75 29,740 131. 4.549 0.0543 132. 12.00 15.00 132. 230. 5.75 133.2 29.740 4.773 0.0538 5.75 29.740 132. 15.00 20.00 132. 240. 133.8 4.773 0.0531 20.00 35.00 132. 132. 245. 5.75 139.0 29.740 4.773 0.0527 132. 5.75 35.00 139.0 29.740 45.00 132. 245. 4.773 0.0527 5.75 45.00 60.00 132. 132. 245. 139.0 29.740 4.773 0.0527 60.00 65.00 132. 132. 250. 5.75 139.3 29.740 4.773 0.0523 65.00 70.00 132. 132. 240. 5.75 138.3 29.740 4.773 0.0531 5.75 70.00 85.00 132. 132. 245. 139.0 29.740 4.773 0.0527 29.740 85.00 90.00 132. 132. 240. 5.75 133.8 4.773 0.0531

FORM 07040 NAVEAC JET DUST SAMPLER TEST 4 UPSTREAM 11APR75 04/22/75

PAGE 3

QSSUM = 326.5924 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
ODSUM = 203.6521 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 274.9590 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAD WEIGHT GAIN = 0.0407 GRAMS

CONC/ACF = 0.0019 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GR/ACF

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-522 PAD2

TOTAL FINAL PAD(S) WEIGHT
TOTAL INITIAL PAD(S) WEIGHT
TOTAL PAD(S) WEIGHT GAIN
1.97430 GRAMS
1.97280 GRAMS
0.00150 GRAMS

BAROMETRIC PRESSURE ABSOLUTE 29.740 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 4 DOWNSTREAM 11APR75 PAGE 5 04/22/75

TIMES	TIME2	DUCT DE	DUCT WE	SAMPLER	DB OPD	SAMPLER WB	PSAH	PPWH	DEHS
0.00	5.00	130.	130.	230.	14.50	136.6	29.740	4.526	0.0540
5.00	10.00	132.	132.	200.	14.50	136.4	29.740	4.773	0.0562
10.00	15.00	132.	132,	200.	14.50	136.4	29.740	4.773	0.0562
15.00	20.00	133.	133.	210.	14.50	137.8	29.740	4.901	0.0553
20.00	25.00	133.	133.	210.	14.59	137.8	29.740	4.901	0.0553
25.00	30.00	133.	133.	220.	14.50	138.4	29.740	4.901	0.0545
30.00	35.00	133.	133.	215.	14.50	138.1	29.740	4.901	0.0549
35.00	40.00	133.	133.	225.	14.50	138.7	29.740	4.901	0.0541
40.00	45.00	133.	133.	230.	14.50	139.0	29.740	4.901	0.0537
45.00	50.00	133.	133.	225.	14.50	138.7	29.740	4.901	0.0541
50.00	55.00	133.	133.	229.	14.50	133.9	29.740	4.901	0.0539
55.00	60.00	133.	133.	235.	14.50		29.740	4.901	0.0534
60.00	65.00	133.	133.	232.	14.50	139.1	29.740	4.901	0.0536
65.00	70.00	133.	133.	230.	14.50	139.0	29.740	4.901	0.0537
70.00	75.00	133.	133.	225.	14.50	138.7	29.740	4.901	0.0541
75.00	80.00	133.	133.	230.	14.50	139.0	29.740	4.901	0.0537
80.00	35.00	133.	133.	230.	14.50	139.0	29.740	4.901	0.0537
85.00	90.00	133.	133.	232.	14.50	139.1	29.740	4.901	0.0536

FORM 07040 NAVERC JET DUS? SANPLER TEST 4 DOWNSTREAM 11AFR75 PAGE 6 04/22/75

QSSUM = 511.6152 TOTAL ACTUAL SAMPLED VOLUME AT SAMFLER TEMP. AND PRESS., ACF QDSUM = 330.2221 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG QDASUM= 444.2592 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

CONCENTRATION DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A SOME-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-525 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.09480 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 2.04480 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.04000 GRAMS

BAROMETRIC PRESSURE ABSOLUTE 29.770 IN HG

FORM C7040 NAVEAC JET DUST SAMPLER TEST 5 UPSTREAM 11APR75 PAGE 2 04/22/75 TIME2 DUCT DR DUCT WE SAMPLER DE OPD SAMPLER WE PSAH **FPWH** DENS TIMES 132. 235. 5.75 29.770 4.773 0.0535 133.5 0,00 3.00 132. 230. 5.75 133.2 29.770 4.773 0.0539 3.00 5.00 132. 132. 5.75 230. 133.2 29.770 9.00 132. 4.773 0.0539 5.00 132. 9.00 19.00 132. 132.1 235. 5.75 138.5 29.770 4.773 0.0535 19.00 30.00 132. 132. 230, 5.75 138.2 29.770 4.773 0.0539 . 5.75 35.00 132. 235. 133.5 29.770 4,773 0.0535 30.00 132. 240. 5.75 35.00 40.00 132. 132. 133.3 29.770 4.773 0.0531 40.00 50.00 132. 102. 235. 5.75 133.5 29.770 4.773 0.0535 50.00 70.00 132. 132. 240. 5,75 138.8 29.770 4.773 0.0531 5.75 75,00 70.00 132. 132. 235. 138.5 29.770 4.773 0.0535 80.00 132. 132. 239. 5.75 139.2 29.770 4.773 0.0539 75.00 5.75 132. 29.770 132. 235. 103.5 4.773 0.0535 30.00 83.00 90.00 132. 132. 230, 5.75 133.2 29.770 4.773 0.0539 85.00

FDRM 07040 NAVFAC JET BUST SAMPLER TEST 5 UPSTREAM 11APR75 PAGE 3 04/22/75

OSSUM = 324.6415 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF ODSUM = 206.8295 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG ODASUM= 276.5499 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAD WEIGHT GAIN = 0.0400 GRAMS

CONCACE = 0.0019 GRAINS PER CUBIC FOOT AT SAMPLER COMDITIONS, GR/ACF

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A ONE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PADI CT-524 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.01220 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 2.00900 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.00320 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.770 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 5 DOWNSTREAM 11APR75 PAGE 5 04/22/75

TIME: TIMES DUCT DB DUCT WB SAMPLER DB DPD SAMPLER WB PSAH PPWH DENS 0.00 5.00 133. 133. 220. 14.50 138.4 29.770 4.901 0.0546 133. 220. 14.50 138.4 29.770 4.901 0.0546 10.00 133. 5.00 14.50 138.7 29.770 10.00 15.00 133. 133. 225, 4.901 0.0542 133. 15.00 20.00 133. 230. 14.50 139.0 29,770 4.901 0.0538 139.6 4.901 0.0538 20.00 25.00 133. 133. 230. 14.50 29.770 23.00 30.00 103. 133. 230. 14.50 139.0 29.770 4.901 0.0538 14.50 4.901 0.0538 30.00 35.00 133. 133. 230. 139.0 29.770 35.00 40.00 133. 133. 230. 14.50 139.0 -29.770 4.901 0.0538 29.770 4.901 0.0538 45.00 14.50 40.00 133. 133. 230. 139.0 45.00 . 50.00 133. 133. 230. 14.50 139.0 29,770 4.901 0.0538 29.770 4.901 0.0533 50.00 55.00 133. 133 230. 14.50 139.0 55.00 60.00 133. 133. 230. 14.50 139.0 29.770 4,901 0,0538 60.00 65.00 133. 133. 232. 14.50 139.1 29.770 4.901 0.0536 139.2 4.901 0.0535 70.00 133. 133. 234. 14.50 29.770 65.00 70.00 75.00 133. 133. 235. 14,50 139.3 29.770 4.901 0.0534 4.901 0.0534 235. 14.50 29.770 75.90 30.00 133. 133. 139.3 80.00 85.00 133. 133. 234. 14.50 139.2 29.770 4.901 0.0535 35.00 90.00 132. 132. 232. 14.50 139.3 29.770 4.773 0.0537

FORM 07040 MAYFAC JET DUST SAMPLER TEST 5 DOWNSTREAM 11APR75 PAGE 6 04/22/75

QSSUM = 514.0999 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS.. ACF
QDSUM = 388.3407 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG

QDASUM= 441.8598 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAGC

TEST DATA

CONCENTRATION, DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-531 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.00440 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 1.95010 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.05430 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.130 IN HG

FORM 07040 NAVEAC JET DUST SAMPLER TEST 6 UPSTREAM 14APP.75 PAGE 2 04/22/75

TIME1 TIME2 DUCT DB DUCT WB SAMPLER DB DPD SAMPLER WB PSAH **PPWH** DENS 5.50 139.4 30.130 4.773 0.0531 0.00 2.00 132. 132. 250. 2.00 3.00 132. 132. 245. 5.50 139.1 30,130 4.773 0.0534 3.00 10.00 132. 240. 5.50 133.9 30.130 4.773 0.0538 132. 132. 5.40 138.2 30.130 4.773 0.0547 10.00 11.00 132. 223. 11.00 14.00 132. 132. 235. 5.50 138.6 30.130 4.773 0.0542 14.00 . 5.50 22.00 132. 132. 240. 138.9 30,130 4.773 0.0539 5.50 22.00 25.00 132. 132. 235. 138.6 30.130 4.773 0.0542 25.00 30.00 132. 230. 5.50 133.3 30.130 4.773 0.0546 132. 30,00 35.00 132. 132. 235. 5.50 133.6 30.130 4.773 0,0542 5.50 132. 240. 138.9 30.130 4.773 0.0533 35.00, 45.00 132. 235. 5.50 133.6 30.130 4.773 0.0542 45.00 65.00 132. 132. 5.50 4.773 0.0546 138.3 30.130 65.00 80.00 132. 132. 230. 5.50 130.6 30.130 4.773 0.0542 80.00 90.00 132. 132. 235.

FDRM C7040 NAVFAC JET DUST SAMPLER TEST 6 UPSTREAM 14APR75 PAGE 3 04/82/75

QSSUM = 315.6220 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
QDSUM = 203.7582 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 268.5752 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAD WEIGHT GAIN = 0.0543 GRAMS

CONC/ACF = 0.0027 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GR/ACF

CONC/DGSCF = 0.0041 GRAINS PER CU FT OF DRY GAS OT 70 DEG.F AND 29.92 IN HG

PAGE 1

TEST DATA

CONCENTRATION DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-530 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT TOTAL PAD(S) WEIGHT GAIN 1.95040 GRAMS 1.94330 GRAMS 0.00710 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 30.130 IN HG

FORM C7040 NAVERC JET DUST SAMPLER TEST 6 DOWNSTREAM 14APR75 PAGE 2 04/23/75

PPWH TIMES DUCT DB DUCT WB SAMPLER DR OPD SAMPLER WB PSAH DENS 0.00 5.00 133. 133. 250. 14.00 140.2 30.130 4.901 0.0530 139.6 30.130 5.00 10.00 133. 133. 240. 14.00 4.901 0.0537 134. 240. 140.4 5.032 0.0536 10.00 15.00 134. 14.00 30.130 140.4 5.032 0.0536 15.00 20.00 134. 134. 240. 14.00 30.130 20.00 25.00 134. 242. 14.00 140.5 30.130 5.032 0.0535 134. 134. 25.00 30.00 134. 242. 14.00 140.5 30.130 5.032 0.0535 134. 14.00 140.5 30.130 5.032 0.0535 30.00 35.00 134. 242. 40.00 134. 242. 14.00 140.5 30.130 5.032 0.0535 35.00 134, 140.4 45.00 240. 14.00 30.130 5.032 0.0536 40.00 134. 134. 45.00 50.00 134. 134. 240. 14.00 140.4 30.130 5.032 0.0536 140.3 30.130 50.00 55.00 134. 134. 233. 14.00 5.032 0.0538 55.00 60.00 135. 135. 238. 14,00 141.1 30.130 5.166 0.0537 60.00 65.00 135. 135. 240. 14.00 141.2 30.130 5.166 0.0535 141.2 5.166 0.0535 65.00 135. 240. 14.00 30.130 70.00 135. 70.00 75.00 134. 134. 240. 14.00 140.4 30.130 5.032 0.0536 140.5 80.00 14.00 30.130 5.032 0.0535 75.00 134. 242. 134. 80.00 35.00 135. 135. 242. 14.00 141.3 30.130 5.166 0.0534 135. 14.00 242. 141.3 30.130 5.166 0.0534 35.00 90.00 135.

FORM C7040 NAVEAC JET DUST SAMPLER TEST 6 DOWNSTREAM 14APR75 PAGE 3 04/23/75

QSSUM = 506.4850 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
QDSUM = 320.8162 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 429.1876 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

4

PAD1 CT-529 PAD2

TOTAL FINAL PAD(S) WEIGHT TOTAL INITIAL PAD(S) WEIGHT TOTAL PAD(S) WEIGHT GAIN

2.02410 GRAMS 1.98650 GRAMS 0.03760 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.980 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 7 UPSTREAM 14APR75 PAGE 2 04/23/75

TIME2 DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH PPWH TIME1 DENS 0.00 1.00 133. 133. 250. 5,50 140.2 29.980 4.901 0.0527 3.00 240. 5.50 139.6 4.901 0.0534 1.00 133. 133. 29.980 3.00 6.00 133. 133. 235. 5.50 139.3 29.980 4.901 0.0538 5.50 9.00 230. 29.980 6.00 133. 133. • 139.0 4,901 0.0542 225. 5.50 29.980 9.00 15.00 133. 133. 138.8 4.901 0.0546 15.00 22.00 230. 5.50 139.0 29.980 4.901 0.0542 133. 133. 225. 22.00 25.00 133. 133. 5.50 138.8 29.980 4.901 0.0546 230. 5.50 25.00 30.00 133. 133. 139.0 29.980 4.901 0.0542 30.00 52.00 133. 133. 24¢. 5.50 139.6 29.980 4.901 0.0534 52.00 55.00 133. 133. 235. 5.50 139.3 29.980 4.901 0.0538 55.00 60.00 133. 133. 240. 5.50 139.6 29.930 4.901 0.0534

FORM C7040 NAVFAC JET DUST SAMPLER TEST 7 UPSTREAM 14APR75 PAGE 3 04/23/75

QSSUM = 211.0631 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
QDSUM = 134.9075 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 180.0978 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.
PAD WEIGHT GAIN = 0.0376 GRAMS
CONC/ACF = 0.0027 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GR/ACF

COMC/HCF = 0.0027 GRHINS PER COBIC FOOT H! SHMPLER CONDITIONS, GRAINS COMC/DGSCF = 0.0043 GRAINS PER CO FT OF DRY GAS AT 70 DEG.F AND 29.92 IN HG

CONCENTRATION.DETERMINATION MADE USING A FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFTCE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-528 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.01680 GRAMS
TOTAL INITIAL PAD(S) WEIGHT GAIN 2.01060 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.00620 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29,980 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 7 DOWNSTREAM 14APR75 PAGE 5

TIMES DUCT DB DUCT WB SAMPLER DB DPD SAMPLER WB PSAH TIME1 PPWH DENS 270. 14.00 0.00 5.00 134. 134. 142.0 29.930 5.032 0.0512 5.00 10.00 134. 134. 220. 14.00 139.3 29.930 5.032 0.0549 15.00 139.3 10.00 220. 14.00 134. 134. 29.980 5.032 0.0549 15.00 20.00 14.00 139.8 29.990 134. 134. • 230. 5.032 0.0541 140.1 20.00 25,00 14.00 134. 134. 235. 29.980 5.032 0.0537 25.00 39.00 135. 135. 235. 14.00 140.9 29.980 5.166 0.0536 30.00 35.00 134. 134. 235. 14.00 140.1 29.980 5.032 0.0537 35.00 40.00 134. 134. 240. 14.00 140.4 29.930 5.032 0.0533 40.00 45.00 134. 134. 240. 14.00 140.4 29.980 5.032 0.0533 45.00 50.00 134. 134. 240. 14.00 140.4 29.980 5.032 0.0533 50.00 55.00 134. 235. 14.00 140.1 29.990 5.032 0.0537 134. 140.4 55.00 60.00 240. 14.00 29.980 5.032 0.0533 134. 134.

FORM C7040 NAVFAC JET DUST SAMPLER TEST 7 DOWNSTREAM 14APR75 PAGE 6 04/23/75

QSSUM = 337.4626 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND RESS. ACF
QDSUM = 213.9625 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 287.7871 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.
PAD WEIGHT GAIN = 0.0062 GRAMS
COMC/ACF = 0.0003 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS GR/ACF

CONC/DESCF = 0.0004 GRAINS PER CU FT OF DRY GAS AT 70 DEG.F AND 29.92 IN HG

ST SUCK SCHOOLSTING OF WHIST BET

CONCENTRATION.DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

TELEVISION OF THE STATE OF THE

PAD1 CT-527 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.06620 GRAMS
TOTAL INITIAL PAD(S) WEIGHT GAIN 2.04540 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.02080 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.990 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 8 UPSTREAM 15APR75 PAGE 2 04/23/75

TIME1 TIME2 DUCT DB DUCT WB SAMPLER DB DPD SAMPLER WB PSAH DENS 235. 5.50 137.8 29.990 0.00 1.00 131. 4.648 0.0540 131. 131. 131. 230. 5.50 137.5 29.990 4.648 0.0544 1.00 2.00 5.50 29.990 131. 136.9 2.00 5.00 131, 220. 4.648 0.0552 220. 5.50 136.9 29.990 5.00 10.00 131. 131. 4.648 0.0552 220, 29,990 5.50 136.9 10.00 20.00 131. 131. 4.648 0.0552 5.50 29.990 20.00 25.00 131. 131. 225. 137.2 4.648 0.0548 5.50 25.00 40.00 131. 131. 230. 137.5 29.990 4.648 0.0544 5.50 137.5 40.00 131. 230. 29.990 4.648 0.0544 52.00 131. 137.2 5.50 29.990 52.00 55.00 131. 131. 225. 4.648 0.0548 55.00 60.00 131. 131. 230. 5.50 137.5 29.990 4.648 0.0544

FORM 07040 NAVEAC JET DUST SAMPLER TEST 8 UPS) REAM 15APR75 PAGE 3' 04/23/75

QSSUM = 209.4048 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
QDSUM = 136.9256 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 180.2832 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

PAD WEIGHT GAIN = 0.0208 GRAMS

CONC/ACF = 0.0015 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GR/ACF

CONC/DGSCF = 0.0023 GRAINS PER CU FT OF DRY GAS AT 70 DEG.F AND 29.92 IN HG

CONC/DUCT ACF = 0.0018 GRAINS PER CUBIC FOOT AT DUCT CONDITIONS, GR/DUCT ACF

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-526 PAD2

TDTAL FINAL PAD(S) WEIGHT 2.05100 GRAMS
TDTAL INITIAL PAD(S) WEIGHT 2.04440 GRAMS
TDTAL PAD(S) WEIGHT GAIN 0.00660 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29.990 IN HG

FORM C7040 NAVFAC JET DUST SAMPLER TEST 8 DOWNSTREAM 15APR75 PAGE 2 04/23/75

TIME1	TIME2	DUCT DB	DUCK MB	SAMPLER	DB DPD	SAMPLER WE	PSAH	PPWH	DENS
0.00	5.00	132.	132.	250.	14.00	139.4	29.990	4 770	0.0528
5.00	10.00	132.	132.	210.	14.00	137.1	29.990	4.773	0.0559
10.00	15.00	132.	132.	210.	14.00	137.1	29.990	4.773	0.0559
15.00	20.00	132.	132	217.	14.00	137.5	29.990	4.773	0.0553
20.00	25.00	132.	132.	220.	14.00	137.7	29.990	4.773	0.0551
25.00	30.00	132.	132.	225.	14.00	138.0	29.990	4.773	0.0547
30.00	35.00	132.	132.	232.	14.00	138.4	29.990	4.773	0.0542
35. აი	40.00	132.	132.	235.	14.00	138.5	29.990	4.773	0.0539
40.00	45.00	132.	132.	235.	14.00	138.5	29.990	4.773	0.0539
45.00	50.00	132.	132.	240.	14.00	133.8	29.990	4.773	0.0536
50.00	55.00	132.	132.	240.	14.00	138.8	29.990	4.773	0.0536
55.00	60.00	132.	132.	235.	14.00	138.5	29.990	4.773	0.0539

FORM C7040 NAVFAC JET DUST SAMPLER TEST 8 DOWNSTREAM 15APR75 PAGE 3 04/23/75

QSSUM = 334.9960 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF
QDSUM = 217.1509 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG
QDASUM= 287.8173 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.
PAD WEIGHT GAIN = 0.0066 GRAMS
CDNC/ACF = 0.0003 GRAINS PER CUBIC FOOT AT SAMPLER CONDITIONS, GR/ACF

PEP-702 24 July 1975

Performance Evaluation
Conducted on a Two-Stage Electrocell Unit
Jet Engine Test Cell
Naval Air Station
Naval Air Rework Facility
Jacksonville, Florida

REPORT NO. 2

PURPOSE:

On July 14, 1975 a field trip was made to the Naval Air Station located in Jacksonville, Florida. The purpose of this trip was to complete the final performance tests on the two-stage Electrocell system installed on this facility's jet engine test cell. This series of tests was conducted to determine if any basic change in the systems' collection efficiency had occurred since the May, 1975 test program reported in PEP-702.

BACKGROUND:

Upon arrival at the test site, a quick inspection revealed the cells were dirty, therefore requiring a wash cycle prior to AAF tests. Upon completion of this wash (accomplished by use of a new prote-type washing control installed by Mr. Doug Pfeiffer of AAF) the system was deemed ready for efficiency testing.

TESTS & RESULTS:

A series of two tests were conducted on the system utilizing the same equipment, procedures and techniques as employed during the May, 1975 program. However these tests were conducted at one system volume only, instead of several various flows as was previously the case.

The system air volume was set at 7334.1 CFM utilizing the 36 square inch inlet plate which provides a cell velocity of 501 FPM. With all other factors the same as the May tests, this recent series provided inlet concentrations of 0.0012162 and 0.0022618 grains per cubic foot for tests 1 and 2 versus 0.0001766 and 0.0001612 grains per cubic foot for the respective outlets. The values provide an average collection efficiency of 89.1% very close to the original test average of 90.4%. (Test data are recorded in Table 1, and followed by computer print outs.)

CONCLUSIONS:

Since the basic purpose was to determine changes or alterations in the ECU's performance at the selected velocity of 501 FPM, these test results provide evidence of the system's continuous efficient operation at this velocity.

RICHARD P. WILLIAMS

RPW:ws

P. Rayner

K. Vestlin

J. Wiegel

C. Bressoud

32<

TABLE 1

Date of Test	7/15/75	7/15/75
Test Number	1	2
Actual Test Time	9:32-10:22	10:59-11:59
Orifice Size, in ²	36	36
Engine RPM (avg.)	7003	7000
Thrust (avg.)	7320	7355
Fuel Flow	6058	6061
E.G.T. (avg.)	992	987
Ambient Temp °F	81	83
Inlet Duct Temp °F	120-132	132
Barometric Pressure, in Hg	30.08	30.08
Total Air Volume , CFM	7334.1	7334.1
Duct Velocity, FPM	733.4	733.4
Cell Velocity, FPM	501.3	501.3
Inlet Conc, grs/ft ³	0.0012162	0.0022618
Outlet Conc, grs/ft3	0.0001766	0.0001612
System Efficiency, %	85.48	92.87

-

TEST DATA

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH DRIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-561 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.06990 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 2.05570 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.01420 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29,920 IN HG

FORM C7040 UPSTREAM TEST 1 07/23/75

PAGE 2

TIMES DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH TIME1 PPWH DENS 130. 5.80 0.00 1.00 130. 240. 137.3 29.920 4.526 0.0536 1.00 2.00 131. 131. 235. 5.80 137.7 29.920 4.648 0.0539 2.00 4.00 131. 131. 230. 5.80 137.4 29.920 4.648 0.0543 4.00 6.00 131. 131. 225. 5.80 137.1 29.920 4.648 0.0547 5.80 138.2 6.00 12.00 132. 132. 230. 29,920 4.773 0.0542 132. 230. 138.2 12.00 40.00 132. 5.8029.920 4.773 0.0542 132. 235. 29.920 40.00 45.00 138. 5.80 138.5 4.773 0.0538 45.00 50.00 132. 132. 235. 138.5 20.920 4.773 0.0538 5.80

FORM 07040 UPSTREAM TEST 1 07/83/75

PAGE 3

0.830M = 180.1515 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF 0.000M = 116.1765 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN H6

QDASUM= 154.2685 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

FIVE INCH SAMPLER WITH A - ONE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-562 PAD2

TOTAL FINAL FINAL FINAL PAD(S) WEIGHT 2.05940 GRAMS TOTAL PAD(S) WEIGHT GAIN 0.00320 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29,920 IN HG

FORM 07040 DOWNSTREAM TEST 1 07/23/75 PRSE 2

TIMES DUCT DB DUCT WB SAMPLER DB OPD SAMPLER WB PSAH · PPWH DENS TIME1 0.00240. 138.8 29.920 4.773 0.0534 2.00 132. 132. 14.00 4.773 0.0542 2.00 5.00 132. 132. 230. 14.00 138.2 29.920 235. 14.00 138.5 29.920 4.773 0.0538 5.00 10.00 132. 132. 10.00 12.00 132. 132. 240. 14.00 138.8 29.920 \ 4.773 0.0534 235. 14.00 138.5 29.920 14.773 0.0538: 12.00 15.00 132. 132. 15.00 29.920 4.773 0.0542 20.00 132. 132. 230. 14.00 138.2 20.00 132. 132. 225. 14.00 137.9 29.920 4.773 0.0546 22.00 132. 14.00 138.2 29.920 4.773 0.0542 ***22.**00 25.00 132. 230. 137.9 29.920 4.773 0.0546 /25.00 132. 132. 225. 14.00 30.00 4:773 0.0546 137.9 29.920 132. 225. 14.00 ***30.**00 50.00 132.

₱ORM 07040 DOWNSTREAM TEST 1 07/23/75 PAGE 3

 $_{
m DSUM}$ = 279.4835 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF $_{
m DSUM}$ = 180.6879 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG $_{
m DDASUM}$ = 240.1550 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND PRESS.

17.

CONCENTRATION DETERMINATION MADE USING A
FIVE INCH SAMPLER WITH A DNE-FOURTH INCH ORIFICE

CARRIER GAS: AIR

TEST PAD(S)

PAD1 CT-563 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.10940 GRAMS
TOTAL INITIAL PAD(S) WEIGHT 2.07780 GRAMS
TOTAL PAD(S) WEIGHT GAIN 0.03160 GRAMS

BAROMETRIC PRESSURE ABSOLUTE 29,920 IN HG

FORM C7040 UPSTREAM TEST 2

07/23/75

PAGE 2

TIME1	TIMES	BUCT DB	DUCT WB	SAMPLER	מאם מת	AMPLER W	B PSAH	PPWH	DENS
0.00	1.00	132.	132.	250.	5.80	139.4	29.920	4.773	0.0527
1.00	2.00	132.	132.	260.	5.80	139.9	29.920	4.773	0.0520
2.60	3.00	132.	132.	250.	5.80	139.4	29.920	4.773	0.0527
3.00	5.00	132.	132.	240.	5.80	138.8	29.920	4.773	0.0534
5.00	6.00	132.	132.	230.	5.80	138.2	29.920	4.773	0.0542
6.00	7.00	132.	132.	225.	5.80	137.9	29.920	4.773	0.0546
7.00	30.00	132.	132.	225.	5.80	137.9	29.920	4.773	0.0546
30.00	60.00	132.	132.	225.	5.80	137.9	29.920	4.773	0.0546

FORM 07040 UPSTREAM TEST 2 07/23/75

PAGE 3

OSSUM = 215.5737 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PRESS., ACF ODSUM = 139.7585 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG ODASUM= 185.7551 TOTAL ACTUAL SAMPLED VOLUME HT DUCT TEMP. AND PRESS.

TEST PAD(S)

PAD1 CT-564 PAD2

TOTAL FINAL PAD(S) WEIGHT 2.08010 GRAMS TOTAL INITIAL PAD(S) WEIGHT 2.07660 GRAMS TOTAL PAD(S) WEIGHT GAIN 0.00350 GRAMS

BARDMETRIC PRESSURE ABSOLUTE 29,920 IN HG

FBRN 07040 DOWNSTREAM TEST 2 07/23/75 PRSE 2

ľ	TIME1	TIME2	DUCT DB	DUCT WB	SAMPLER	DB OPD	SAMPLER	WB PSAH	PPWH	DENS
ē	0.00	2.50	132.	132.	250.	14.00	139.4	29.920	4.773	0.0527
•	2.50	2.50	132.	132.	270.	14.00	140.5	29.920	4.773	0.0513
l	2.50	4.00	132.	138.	260.	14.00	139.9	29.920	4.773	0.0520
O	4.00	6.00	132.	132.	250.	14.00	139.4	29.920	4.773	0.0527
.	6.00	7.00	132.	132.	240.	14.00	133.8	29.920	4.773	0.0534
ľ	7.00	9.00	132.	132.	230.	14.00	138.2	29.920	4.773	0.0542
	9.00	11.00	132.	132.	220.	14.00	137.7	29.920	4.773	0.0550
	11.00	15.00	132.	132.	215.	14.00	137.4	29.920	4.773	0.0554
ľ	15.00	25.00	132.	132.	220.	14.00	137.7	29.920		0.0550 🦮
	25.00	30.00	132.	132.	222.	14.00	137.8	29.920		0.0548 %
	30.00	35.00	132.	132.	225.	14.00	137.9	29.920		0.0546 👙
	35.00	42,00	132.	132.	226.	14.00	138.0	29.920		0.0545
	42.00	45.00	132.	132.	225.	14.00	137.9	29.920	4.773	0.0546
	45.00	50.00	132.	132.	226.	14.00	133.0	29,920	4.773	0.0545
_	50.00	\$5.00	132.	132.	227.	14.00	138.1	29.920 ₀	773	0.0544
	50.00 55.00	60.00	132.	132.	226.	14.00	138.0	ເ 29.920ິ	4.773	0.0545

FORM 07040 DOWNSTREAM TEST 2 J07/23/75 PAGE 3

. JM = 334.8472 TOTAL ACTUAL SAMPLED VOLUME AT SAMPLER TEMP. AND PPE\$3.. ACF RDSUM = 217.1904 TOTAL DRY GAS SAMPLED VOLUME AT 70 DEG.F AND 29.92 IN HG RDASUM= 288.6711 TOTAL ACTUAL SAMPLED VOLUME AT DUCT TEMP. AND FRESS.

375

APPENDIX A-2

Air Samples from Electrostatic Precipitators;

Results of;

Naval Air Rework Facility, NAS

Jacksonville, Florida

Delfanso

·--Code 340 16 Jun 1975

MEMORANDUM

 ε_{m_1}

From: Code 340 To: Code 610

Subj: Air samples from Electrostatic Precipitator; results of

Ref: (a) MELR No. 3-74

Encl: J-79 engine data with electrostatic precipitator

J-52 engine data with electrostatic precipitator J-52 engine data without electrostatic precipitator activated

- 1. Air sampling for determining the efficiency of an electrostatic precipitator has been performed on the NARF model as required by United Engineers and Constructors, Incorporated, for SOUDIVNAVFAC.
- 2. The test results are forwarded as enclosures (1) and (2). The following information concerns the data:
- a. The format and calculations are those used in reference (a) in order to allow comparison of similar data.
 - b. All tests were performed at normal rated power.
- c. The model exhaust gas entrance tube was blanked off between the April and June tests.
- d. The precipitator was cleaned in April after the J-79 tests and not thereafter.
- c. The June 4 tests were made with both power packs on the precipitator activated.
 - f. The June 5 tests were made with one power pack nonfunctional.
- g. The June 6 morning and afternoon tests were made with no power to the precipitator.

THOMAS Superintendent

405

J-79 ENGINE DATA WITH ELECTROSTATIC PRECIPITATOR

	INLET	OUTLET	THE	OUTLET	INTE	CUTLET
Date	4-17	4-17	4-18 AM	4-18 AM	4-18 PM	4-18 PM
Thow at #1 Sump gal./min.	9.66		68.6		19.6	
Vol. of dry ges sampled. SCF	92.79	81,66	87.99	83,31	84.27	84.82
Stack flow rate, SCFM, dry	9276.0	8149.4	8404.3	7621.5	8607.6	7920.2
Stack gas velocity, at stack	763.8	665.8	703.8	620.5	698.0	698.3
Concilions, i.p.u.	15.75	15,15	17.13	14.89	16.66	15.39
Stack was temp, degree F.	132	131	132	131	132	131
Isokinetic, %	97.5	97.6	102.0	106.0	95.0	104.3
Particulate Results (a) Probe and filter catch Grains/SGE, drv. x 10-3	6.17	2.90	3,48	1.05	3,14	I.05
(b) Total Catch Grains/SCF, dry, x 10-3	7.30	3.54	3.92	1.20	3.21	1,27
(c) Particulates from #1 Sump Water sample grains/SCF, x 10-3	5.41		5.64		5.14	
Particulate Removal Efficiency Reced on air sample (a), %	53.0		8.69		66.6	
Based on air sample (b), %	51.5		6.49	. •	60.4	
Based on Lotal (air and water, a+c), %	74.9		88.5		87.3	
	72.1		86.8		84.8	
Entrained Water Removal, %	11.5	٠	44.8	•	22.7	

Enclosure (1)

-52 ENGINE DATA KITH ELECTROSTATIC PRECIPITATOR

	INLET	OUTLET	INTEL	00.175.1
	7	7-9	Ţ	6− 5
late slamp gal./min.	10.84	77.942	9.92 80.930	78.660
701. of dry gas sampled, Sur Stack flow mate, SCFN, dry	7552.8	7302.7	8339.9	7409.9
Stack gas velocity, at stack conditions, f.p.m.	682.5	609.9	693.5	611.1
Moisture, % by volume Stack gas temp, degree F.	133	131	132 94.5	131 103,4
Isokinetic, %				
(a) Frobe and filter catch	1.44	2,65	1.26	1.37
(b) Total Catch Crains/SCF, dry, x 10-3	1.91	3.44	1,56	1.92
(c) Particulates from #1 Sump Water sample grains/SCF, x 10-3	1.02		0,97	
Particulate Removal Efficiency Based on air sample (a), %	-84.03		-8.73	
Based on total (air and water, atc), % Based on total (air and water, atc), %			38.56	
Based on total (all and Entrained Water Removal, %	27.34		8.50	•

page 1 of 2 pages

ENGINE DATA WITHOUT ELECTROSTATIC PRECIPITATOR ACTIVA

有情况,是是是一种,他们也是一种,他们也是一种,他们也是一种,他们也是一种的,他们也是一种的,他们也是一种的,他们也是一种,他们也是

	INLET	OUTLET	INTEL	OUTLET
Date	6-6 AM	6-6 AM	6-6 PM	6-6 PM
Flow at #1 Sump gal./min.	10.34		10.34	•
Vol of dry gas sampled, SCF	82,395	79,456	85,549	79.521
Stack flow rate, SCFM, dry	8265.7	7358.0	8270.8	7466.1
Stack gas velocity, at stack conditions, f.p.m.	698.7	614.5	697.3	622.4
Moisture, % by volume	17.24	16.32	16.82	15.98
Stack gas temp, degree F.	133	132	133	132
Isokinetic, %	97.1	105.2	100.8	103.8
Particulate Results (a) Probe and filter catch		,	:	;
Grains/SCF, dry, x 10-3	1.53	1.92	1.49	1.90
(b) <u>Total Catch</u> Grains/SCF, dry, x 10 ⁻³	1.92	2.40	1,99	1.90
(c) Particulates from #1 Sump Water sample grains/SCF, x 10-3	1.02		0.93	
Particulate Removal Efficiency				1
Based on air sample (a), %	-25.49		-27.52	
Based on air sample (b), %	-25.00		-19.60	•
Based on .tal (air and water, atc), %	24.71		21.49	
Based on total (air and water, b+c), %	18.37		18.49	
Entrained Water Nemoval, %	15.60	••	15.37	-
				,

Page 2 of 2 pages

Enclosure (2)

V.

GENERAL COMPUTATION SHEET Inited engineers & constructors inc.

NAME OF COMPANY NAVFAC PROTOTIDE KEED UNIT'S
SUBJECT COIRTECTION OF MARF-JAX DATE OF MARF.

CALC. SET NO.

PRELIM.

FINAL

VOID

SHEET OF

J.O. 6/83-003

Pey COMP. BY CHK'D BY

O DATE

DATE

DATE

- NATER CALC. OF STACK Fluent ApproxIS

to be based on 10 FTZ duct instead

OF 10 FTZ STACK

10 Term OF 4-17 / what

VEL - 763.8 FRILL

ACFUL USING 1674 - strek = (763.8)(16) = 17,220.8 AcFUL

Day ACFUL C 15.7584,0= (12,720.8)(1-1575), 10,296 DACFUL

Try SCFUL C 70°F - (10,296)(530°R) - 9217 SCFUL, day

(4604132)

REport Lists 9276, Ullerince is mobilely the to state meri.

- Actual duct dimension ARE 4 × 2-6 10 FTZ : Strick Fluorite Juneaurium should be connected by 10/16 = 0.625
- Data LICTED UNDER Particulate results (c) (Particulates From +1 sup) ARE Calculated As Ellaw

Conc = (GR/(IL) (GAL/MIN)

435

GENERAL COMPUTATION SHEET united engineers & constructors inc.

NAME OF COMPANY LANTAC PROLITION WHITES
SUBJECT CORRECTION OF MORE - LAX DAT OF DUCT SIZE

connected by 16/0 = 1.6

	CAL	.C. SI	ET NO.
PR	ELIM.		
FIN	AL.		
vo	۵		
SH	EET		OF
J, C). (0	163-	W 5
R _E y	COM	P. BY	CHK'D BY
C	Da		DATE
	DATE		DATE

- Data LISTED WIDER PORTURALITE TREMONDEPPRIERRY
NURT DE COMPATES TO REFLECT higher carentontans

- Data Connected As Follows.

JTACT	4-17	4-18 Aud	416 Pm	6-4	1	6-6 Aus	
Locations	iuler	iulet	ndet	Iulct	121.2	nukt	nulet
STACK VEL - REport	9276	8404	86 08	-7953	8340	eruc	8271
Court.	5798	52\$3	5380	4971	5713	5166	5169
PART FORM - KEPORT - CORR		 .00564 .0090Z	.0514 .0812	.00163	.00097 .00188	.00167	: .,a2093 1,40/49
Littleford basedar Hill + untend b+c) - Report - Conn.	77.1	€6.6 90.73	84.8 88.89	- \7.4] 2,82.	24.11 33.52	1	18.49
				[] !		1	

APPENDIX A-3

Basis for Operating Cost Computations

FEMASIAS T JUILLY

	Record No. 6.103.003.								
		ENGINE?	PARAMETE	ıs.		TEST COLL	. PARAM	eters.	
		FOWER LYL	ENGINE FXh And	Estaves Exh Temp		AUG	CELL	CELL Exh Temp	4
			16/sec	• =		-	16/SEC.	υF	
	J 79 ENGINE	IDLE	50 \70	350		.5 .9	75 373	275	
	w/AB	751 MAN	170	1150		1.6	340	625	<u>-</u>
		AB	170	3100		1.2	374	1550	
			1 13				 	1	
		t						1	
								1	
									, inchine
		 	<u> </u>					 	
		 	 	<u> </u>			 		
1		 	 				<u> </u>		
		 	 	 -			 -	 	
L		 -	 	<u> </u>			<u> </u>	 	
		 	 				<u> </u>	 	
· }									nfished .
<u> </u>			İ						Heliki
									and a
									g.
			ļ						
ļ								ļ	
<u> </u>	35037		\			<u> </u>			. "
	350 16/ SEC EMBILE	IDLE	130	350	ļ	1	260	715	
	w/AB	70%	240 350	500 650		1	480	797	
		AF3	350	3500		1.2	770	360 (690	
' 							'''	10,0	
						,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,			
<u> </u>									
		<u> </u>	ļ		<u> </u>				
-			 				 		
		}	 	 		 		 	
<u> </u>		 	 	 	<u> </u>		 	 	
1		1			1		 	†	
		ļ							
							 		
	н	1	1	1	1	1	1	1	l

METERS FOR TYPIN) AE TEST CYCLE

						م خرجه مساوحته		-					· • · · · · · · · · · · · · · · · · · ·
			1715 cools					GA3 Cleaning					
	l	EVADORUTE TAC CO	TEMP	SAT EAD Flue	ATTANCGAVZ ZTAY	COLALITE RANGELLE BOR		PRECID DEMAND		TA EMIT SIYUMT	CoulouT Tumped	THAKES GNUE	OUGSTAN!
T		PH29/16 CELL	<i>با</i> د	ACFW	GALL	azur		KW		min	GAL	CAL	GAL
7		Fluid											
7		0,04	100	67,500	22	50		23		50	1100	2500	1400
T		0.12	130	340,000	280	350		26		30	8400	10,500	2100
m I		0135	132	362,000	331	425		25		30	9900	12,750	2850
\prod		0.35	164	547, 200	943	۱۲۵۵		28		10	9430	12,000	2570
										Nomia	37,760	28630	8920
m I													
\rfloor													
1													
1				<u> </u>									
4		·											
4													
Ц			ļ										
4													
4		0.03	89	225,00	56	70				55	3850	3080	770
Ц		0.05	105	440,00	173	220			i	30	6600	5200	1400
4		0.07	112	666,000	3 \$ 5	450				30	/3500	10,950	2550
4		0.36	167	1/75,000	2053	2600				72	57,260	45,200	12,000
_			ļ							137mu	81,100	64430	16,726
_			 										
4													
4			 	 	 	 							
4				 		 							
_			 	 								 	
_				-	ļ	ļ		 				 	
_			 	 	<u> </u>					ļ. —			
-					 				<u> </u>		<u> </u>	 	
-		 -	1	 	 	 					 	 	
-													
			†			 					<u> </u>	——	
-	-		 	 		 					†		
_		 	 	 	 	 					†	<u> </u>	
-			†	 		 		<u> </u>	·	<u> </u>	<u> </u>	 	
-			1	<u> </u>	-						سيضي سيب		
	•	•	•	•	•	ı	'	1	1	1	•	•	•

Form	5016	flev.	11/74

GENERAL COMPUTATION SHEET united engineers a constructors inc

NAME OF C	OMPANY MAYFAS Trolitized Til 17 UNITE	
	COST OF FRESIDITATION ENGINEETTICAL	

	CALC. SET NO.						
PR	PRELIM.						
FIN	AL						
vo	O						
SH	EET	/	OF /				
J.C	ام) .	33-	003)				
"EV	COMI	P. BY	CHK'D BY				
0	DATE 7-9		DAYE				
	DATE		DATE				

IL GOODON AND LUMB

Corrects of S nounter

carryundele renignas so celle (sunse x 7 high)

Early hoursouted TRAN (8 WINE) STITUTE by ONE LG-8

TREWERL PACK. TRATTUG 6.5 A @ 170 VAC, 500 W

system will operate it seating Ders DAF

Kinz (7 Hars/ produle) (E model E) (5 Km/Rus) = 28 Km

For 170 min test cycle - 28 her thorz 56 Kuh 6.03/Kuh- \$ 1.6E

I 1,700,000 ACTUL ... T

consists of 16 modules Each mode indian 1 64 cous (500) EACH ROW Suppliers I , said LG. & Theren Dr. K

Kw=(8:1201/2014le)(1600 hals)(.5160/120) - 64 Kw

Line 137 man TEST agala - 64 km. 187/60= 146 kauh (c 103/Kuh - 1438

47<

IDISCIPLINE

United engineers & constructors inc.

NAME OF COMPANY MAYEAC PROTOTION PORCED	
TOTAL CANDOLLINE MILLIA	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,

CALC. SET NO.						
Pf	ELIM.					
FI	NAL					
VC	010					
St	EET	1	OF I			
J.	0.61	83	.003			
*Ey			CHK,D B.	~		
0	DAVE 7-4	M 1.75	DAYE	-		
	DATE		DATE	7		

I witer wirempton

500,000 ACFM - 28,830 GAL

II. Paupinis Paven

300' HD, 60% Purp EFF, 90% motion EFF. Kuh/10006m = (0.00315)(300)(1) = 1.76

500,000 ACFM - 37,750 CM Purpel Kuh = (37.75) (1.76) . 66.44

1200,000 ACFUT - 81,100 GIL Tempel Kuh = (81.1)(1.76) = 142.74

$\overline{\mathbf{II}}$.	Samo			
	item	ausunption	we.	COST
	waters	20030 CAL 66.44 Kuh	-35/1000 GR	10.09
	TOTAL			112.08
	Pewer.	64430 147.74	35) much "05/ Enh	27 59
	चार (485		3 2LR2

GENERAL COMPUTATION SHEET United engineers & constructors inc.

NAME	OF	COMPAN	· MAUFEL	118/	hype Kra	<u> гар</u>	UNIT/S
SUBJE	CT	(057	of Oversa	ia (155-C/-m	HOILA	

CALC. SET NO.						
PRE	LIM.					
FIN	AL					
VOI	D					
SH	EET)	OF (
J.C)	, í	J3			
MEY	COM	P. BY	CHK'D BY			
0	DATE 7-4		DATE			
	DATE		DATE			

I. Rimping Costs

35 HD, 50% rupert, 85% while EPF

(.5) (.25) (0.259

500000 ACFUL - 8970 and Rimpers

Kalo (1.259) (892) 231

1300, an Action - large on Pared

Wike 1259) (1672) = 433

Reproduced from best available copy.

GENERAL COMPUTATION HEET ATTENDED TO THE TOTAL COMPUTATION OF THE TOTAL

			<u> Muctiali</u>				
	CALC. SET NO IUN						
PRE	PRELIM.						
FIN	FINAL						
VOI	D						
SH	SHEET \ OF 4						
J.C	ام .	83-6	いづ				
REV	COM	P. 8Y	CHK,D BA				
0	1)4	11_					
	DATE	1	DATE				
	DATE		DATE				

I. WATER CONTLINIPTION JURILIA WASh CYCLE

NAME OF COMPANY MARTACLASTIC UNITES UNITES

SUBJECT COST OF WAYDING PRECIDITION

- Each westiers Assembly consumes 15 GAM
 - when in drawing
- WASher Horses tremel GFT/min vertically; Associations
 TRADEL GTT/min horsessatilly
- were noted water 4 ventical masses; issentitles (400 units) then more horizonally

500000 KFUL TOTAL TIME VE

Verstically - 1277 hich worded / 6 F7/mer x 4 DASSES

MIN 34 = NEXA SOLUTO A SOLUTION AS X

horizontelly = 24 wice/40000 mm 6 min

TOTAL - JATOT

MORRIES OPERATE ON UNITED VEILTY, (PASSES

Consimption = 15 Gont/Assy x B ASS/ x 48 min = 5760 GALLONS

1.800,000 ACTIM - TOTAl Think - Veretrilly - 13FT hich mould GFT/min & 4 Parses

X 24 WHO / 4 WW NO ASSILE 57 MILL

horrizontal (

6 mil

Reproduced from best available copy.

Toul

58 MILY

GENERAL COMPUTATION SHEET united engineers & constructors inc.

NAME OF COMPANY MANTAC TOWNSHIP TREETS UNITIES	
SUBJECT CONT OF WAShirly Precipilitar	

Consumption - 15 Gray Assept 16 Assept

		_	والمتحدد وال		
CALC. SET NO.					
PRE	LIM.				
FIN	AL				
VOI	0				
SH	EET	2	OF 4		
J.C). 6	83.	ロロラ		
REV	COM	P. BY	CHK'D BY		
0	DATE 1-4		DATE		
	DATE		DATE		

II. EMERCELLICET OF WASTER DRIVES

16 17 draws motores operationaline 54 min (1 times) times

500,000 Actal - House Chis . 0.15 KW x & modela & 54 mm = 1.09 Kuh

Mangrow MEFFET - Facer 18115 - 0.15 Kmx 16 mobile x 58 mm = 2.33 Kuh

III ENERGY COST OF MAYOR Wash witer

Kuly/1200 GAL = (0.00015, (1)) (State) (Hapert) (motor 199)

SWOWN ACTULE 5760 GALL

Kulyramine (0.03/5)(100')(1) = 0.618

Kush= (618) (5.7%). 3.5

1200 an Actual 17,500 and

Kuh/1000 Ent - (0.00315)(140)(1) = 0.866

Reproduced from best available copy.

Kuh- (0.804)(165). 11.8

GENERAL COMPUTATION SHEET United engineers & constructors inc.

CALC. SET NO.						
PRELIM.						
FIN	AL					
VO	D					
зн	EET	3	of 4			
J.C). 6	63-	003			
NE^	COM	P. 8Y	CHK'D BY			
0	DATE 7-4		DATE			
	DATE		DATE			

NAME OF COMPANY TO THE TALL WITTER UNITES

IL VUTSING & CONSUMPTION

channe & of 4 vanish (15202 -

Grandent pion = 5760 all + 1/2 aide - 40 = 77 dal

1,000,000 Miles Cardin plan 12500 MI x 1/2 cepte x 40 = 156.75 cht.

VAILABLE TO DDC DOES NOT FULLY LEGICLE PRODUCTION

	4.161,19.1.1	``\	,		, p
	item	Secretary from	1	. T	<u> </u>
(ب	walers	5800 UN	*35 /1000 SAC	* 2.03	AVAILABLE
279	determine	72 EAL	"LOUJUALL	¹ 285	YYDO
	Manuel.	4.59 Kid.	Sus/Kill	10.14	3
•	Total			1 290,17	Verz carphtiansh
ત્રી	witer	12,500 00	14.35/1000 6nd	4.36	
25/41	Jorovent	156 val.	1 40/011	\$624	 - -
N K	KIDOR	14.13 Kill	The King	*0.4Z	!
	1157	Reproduced from best available copy		\$ 475.80	recomplete wish

GENERAL COMPUTATION SHEET	_
united engineers a constructors inc.	CALC. SE
and the constructors inc.	PRELIM.
	FINAL

NAME OF	COMPANY NAVEA C PINTOLIPE UNIT/S
SUBJECT	Cost OF WAShing Precipitator

Estimate of Priceys " with schedule 1ST Limik - CHE FOR IL - (10 TENS) 200 hours - weeke no (40 tax)

	CAL	.C. SI	ET NO.
PR	ELIM.		
FIN	AL		
VO	O		
SH	EET	4	OF 4
J.C	0.61	83.0	17'S
REV	COM	P. BY	CHK'D BY
0	1) d. 0AYE 7-4-		DATE
	DATE		DATE

Numbers of tery per audit with

1 + (3)(2) - 21/2 wrehin Pin do Ferri - 16

COST POR TEST

50,000 ACFULL - \$290,17/16 : \$18.14/ TENT 120, as ACPACL - \$1.1880/16 = \$39.30/ TEST

Form 8015 Hav. 11/74

(DISCIPLINE)

į, i

GENERAL COMPUTATION SHEET United engineers & constructors inc.

NAME OF COMPANY NAVEAC PROJECT PRESSURE FILTER SYSTEM

I. MAXIMUM TOTAL PARTICULATES VELLETATED

- (1) Estimate Just they ken Environ
- (8) Use Eurissian Incloses frankepont

	CAL	.C. SI	ET NO.
PR	ELIM.		
FIN	AL		
VO	O		
SH	EET	1	0F 4
J.C	0.618	33-0	03
#EV	COM	P. BY	CHK,D BA
0	126 001 - 11		DATE
	DATE		DATE

- 179 EMGINE IM 180 mill apple.

House Ly L	thaist	ブSF C	Fuel. Flaw	English	Time (Printed 106.RIG	ibs Pont cera
IDLE	410 li	3.75	1620 mh	1.167 lb/1685	50 mill	25	42.5
ILIT	7300	.82	6000 WI	1.0	30 min	55.5	55.5
wiil.	10,000 lb	.86	4375 m	10 "	30 must	867	86.7
AB	17,au 11.	1.93	32,800 00	.4	10 mint	101.2	40.5
	•	:	1				225.2 16s

- 350 16/5EC ENGINE IN 137 mm cycle

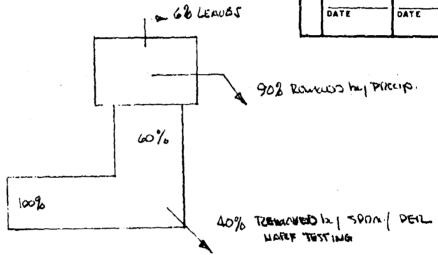
		, 1	1		Ì		
IDLE	900	1.6	1440 mh	1.2	55 mid	24.79	29.1
INT	9,300	.6	5580	.5	30 *	51.34	25.7
mil	19,400	.65	12,610*	.5	30 .	116.01	58
AB	29,400	3	88,20c	.3	22 •	595.04	178.5
							291.3 lbs Per 1547
		ļ		سة الأرا		}	

GENERAL COMPUTATION SHEET United engineers & constructors inc.

NAME OF COMPANY MAD FAC MA	WINE HELL	UNIT/S
SUBJECT OPERATING COST	Riedura Filtre	/sta

CALC. SET NO.				
PRI	ELIM.			
FIN	AL			
VOI	٥			
SH	EET	7	OF 4	
J.C	ایک .(£3-0	-3	
RE'A	COMI	P. BY	CHK'D BY	
0	DATE		DATE	
	DATE		DATE	

II. Distrabilition of Particulates



TOTAL MOILITALITS CONTENTED PERS TEST.

179- 225.2 x.94 · 211.7 lbs

35016/sec - 7913 1.94 - 273.8 165

III. OPERATING COST FACTORS (EXCLUDING PUMPING)

- 1) FILTER AID IS ADDED AT A RENTED between 0.1 and 1.0 16/16
 PRATICULATE CHRISHIP FILTERING
- 2) Filters is Precontens with Silter AID After Environ chamber (0.1 115/soft)

NAME OF COMPANY MATTAC MITTING UNITIS	
SUBJECT OPERATING COSTS of PRESSURE FILTER Systems	

- 4, Clearing Requires Apply 10,000 SCF OF AIR FOR CRIMING (800 SPFT UNIT)
- 5, Cleaning requires Approx 3-5 SCFM

 For Approx 15 min to operate leaf vibrators

CALC. SET NO.

PRELIM.

FINAL

VOID

SHEET 3 OF 4
J.O. (p183-003

Tey COMP. BY CHK'D BY

O DATE

DATE

DATE

DATE

ole villingligetten om - vikkimanjinglikustur ostillusillikistillisillisillisillisillisilmillisilmi

USING 800 SOFF FILTER, A TOTAL OF APPENX 1400 Ibs PART
COULD be collected between cleanings barren on additional ib Filling
mic/Ib part.

THEIR CLEARING : TOTAL FITTER AND ADDES with swiming = 1400 lbs

1480 lbs @ 20/11/

TOTAL AIR to drug = 10,000 SCF

AIR to VIDRATE. 5 SCHI'IBMA = 75 SCF MEDLECT

541 10 au 56 6 20/10016

+ 187/ Cheannas

For 179, Edtermail DE demies lance Every 10-12 Eaty ouce Prient to and following many water

For 356 16/300, Think EVERY 8-10 TOXI

dem	6016	Rev.	11/74

GENERAL COMPUTATION SHEET

United engineers & constructors inc.

NAME OF COMPANY LAPLACE	estatifica Kaisage UNITES
SUBJECT LAND LEVEL CO.	Ressure Filter System:

CALC. SET NO.

PRELIM.

FINAL

VOID

SHEET 4 OF 4

J.O. 4183-123

Tey COMP. BY CHK'D BY

O DATE

TOTE

DATE

TOTE

DATE

IV. Hampaug Costs (ON PER TEST BASIS)

ALG Paysholo 100" 11 125 (1.60) 10 1/1.000

Kich / 100 GAL = (0.00317)(10)(1) . 0.513

500,000 ACFINA - 8470 611 / 1587

Kuh- (0.618) (842) - 5.51

1,200,000 ACFU

Kuh = (0018) (16:12) = 10:33

Reproduced from best available copy.

COPY AVAILABLE TO DDG DOES NOT PERMIT FULLY LEGIBLE PRODUCTION

APPENDIX A-4

Conversion Test Data to
Emission Factors

Form 5016 Mev. 11/	/74
--------------------	-----

GENERAL COMPUTATION SHEET United engineers & constructors inc.

NAME OF	COMPANY MAYE	AC Prototype	Precio	UNIT/5
SUBJECT	(Suversions o	E TEST DATA TO	1/05/100 BK	

CALC. SET NO.				
PRELIM.				
FIN	AL			
VOI	D			
SH	SHEET \ OF 4			
J.C	J.O. GIE'S-W3			
REV	COMP. BY		CHK'D BY	
	De	 	Da	
O DATE 7-3-7Y		DATE		
1	DATE	•	DATE	
•			.	

I. FUEL HEAT INPUT

AVERAGE ENGINE FUEL Flow Union TERTS - 6000 160/HR
FUEL HEATING VALUE - 18,400 Bty/16
FUEL HEATING VALUE - (18,400) (6000) = 110.4 × 100 TSIG/HR

II. TOTAL GAS FLOW

- ESTIMATED ENGINE AIR MOST TION & TAURIME 180/b/sec
- FUEL Flux + 6000 114/11/ 3602 Sec/AIR 16716/SEC
- Total ENhouse Medicts = 181.67 16/sec
- with ANDERLAGE TEXT pasteries to loss And saturation temp
- TOTAL CELL FLOW = (1. + 09) (1817) = 345.7 16/SEC

III Tord Tring CELL Flow

- moistaile in FUEL

ASSEMB 10% He we had; 0% FREE HEO
1125 HEO/HE = 3.94

Form 5015 Rev. 11/74

(DISCIPLINE)

GENERAL COMPUTATION SHEET United engineers a constructors inc

CALC. SET NO.				
PR	LIM.			
FIN	FINAL			
VO	ם			
зн	SHEET 2 OF 4			
J.C	1.0. 4183-22			
"EV	COMP. E	34	СНК	O BY
0	106m	, , ,	DATE	
	DAYE		DATE	

NAME OF COMPANY MANTAC PROLIT DE VICEITO UNITIS

Total Het merching = (0,844 Harring/Hamil)
(6000 Har/ma) (/3000 Harring) 1.49 Hoto/see

- moisting in incertes and anomorto AIR

Total AIR: 345.2 - 1.67 - 343.5 lb/sec

Assume in air 16 Ho/ 16 chy AIR

Total day Air: 343.5/1.012 2 339.216/sec

TRATE MAR MORETURES: 343.5-339.7 = 4.3 16/502

- Total morelines : 1.49 + 4.3 = 5.79 lb/sec

Total morelines : 1.49 + 4.3 = 5.79 lb/sec

Total day once 345.2 - 5.79 = 339.4 lb/sec

downly / Uniquese 600 F, 29112 44 = 0.07636 lb/ff3

Total volume = (339.4 lb/sec X 60 sec/min) (/07636 lb/pscf)

= 266.692 Day SCFM

Reproduced from best available copy.

	5015	0-4	11/74
Form	5015	Hev.	1 1//-

	GENERAL COMPUTATION SHEET		
ė	united	engineers	& constructors inc

CALC. SET NO.				
PRE	LIM.		1111	
FIN	AL			
VOI	0			
SH	SHEET 3 OF 4			
J.C	J.O. 6183.003			
ME _V	COM	P. BY	CHK'D BY	
0	0 DATE 7-3.71		DATE	
	DATE		DATE	

NAME OF COMPANY NAVFAC PROPUL POS PIRCLES UNITES SUBJECT CONVERSION OF VEST DATA TO MON TOUR

IV RECATIONShip DETWEER ON DOCF and IbylouBty

LET A = GR/DSGF (from trap data)

165 Peneticulate/HR = (A GR/DSCF) (/7000 16/412) (266,692 DSCF/min.) (60 mis/412) = 2286 A lbs/HTZ

20.71 A 163 PAUT/104 Btu = 2286 A 16x/412 = 110.4 100 Bt /HIL

I Dut Concresion

MARKETER IN ET CONCENTRATION WERENDERD IN THIS A' FESTE TESTE = 0.0045 GR/ DICE (4-9-77) , MILLIMUM ... 0.0018 Majoristi - (0045)(20.711)= 0.093

- maximum outlet concentration recorded in type 'A' Texts. TEARS = Q. 0008 GR/DSIF (4.9.11) MILIMALINA GOCOL

165/100 Ble - (0.0008) (20.71) - .017

0.002

61<

- MAKINGUM INJET CHICENTRATION RECORDED IN TELES B'

TOTS = 0.0073 + 0.00866 = 0.01596 GZ DSCF (4-17-75); HYIMIAN +

Form 8015 Rev. 11/74

(DISCIPLINE)

GENERAL COMPUTATION SHEET

United engineers & constructors inc.

NAME OF COMPANY MAYE	AC PIDTOTIDE	Priccip units
SUBJECT CHURSSINH O	Teat Dita	to 160/104 15th

CALC. SET NO.				
PRI	PRELIM.			
FIN	AL			
VO	D			
SH	EET	4	0F 4.	
J.O. 6183-003				
E	COM	P. BY	CHK'D BY	
0	D'S PATE T	un 1-71	DATE	
	DATE		DATE	

-maximine millet concentration resconded in Type 3' lero

= 0.00354 62/05UF

MILLING. - 0.00120

161/164 Kt = (0.00354) (2071) - 0.073

0.025

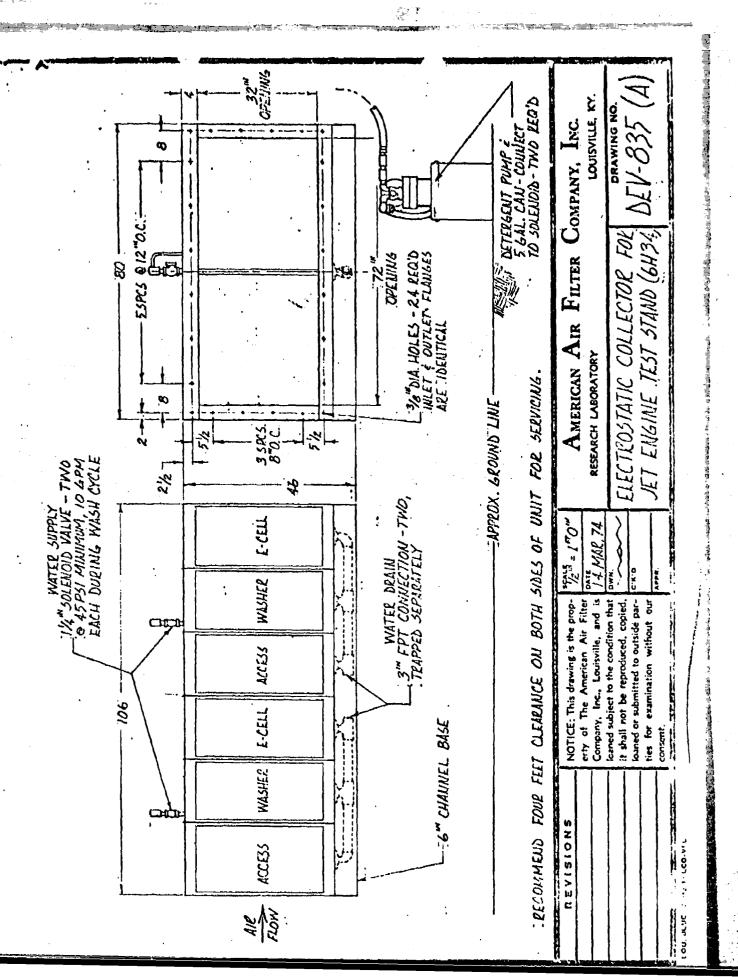
APPENDIX A-5

Dimensional Drawing

of

Two-Stage

Electrostatic Precipitator



SECTION 7

BIBLIOGRAPHY

- A Study of Means for Abatement of Air Pollution Caused by Jet Engine Test Facilities - United Engineers & Constructors Inc. - August 1973.
- 2. Jet Engine Test Cell Pollution Abatement Efficiency Tests Naval Air Rework Facility, NAS, Jacksonville, Florida, March 1973 May 1974.
- 3. Standards of Performance for New Stationary Sources Federal Register, Vol. 35, No. 247, December 23, 1971.
- 4. Procedure for Two-Stage Precipitator Test Program United Engineers & Constructors Inc. May 1974
- 5. The Effect of a Model Electrostatic Precipitator on Particulate Emissions from a Gas Turbine Engine Test Cell Naval Environmental Protection Support Service, Aircraft Environmental Support Office, NARF North Island, California Report No. AESO 111-75-9, October, 1975.